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SST Technology Follow-On Program — Phase I

COMPATIBILITY OF SST MATERIALS WITH TITANIUM ALLOYS; VOLUME II, MANUFACTURING—AID MATERIALS

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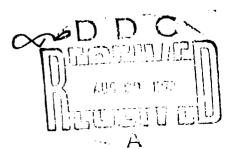
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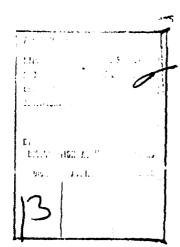
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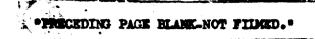


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16. Abstract This document presents data on the compatibility of titanium Ti-6Al-4V with manufacturing aid materials, which might have fabrication of the U.S.A. supergonic transport. Testing procedures are tabulated.	ve contacted titanium during
Test methods included use of a simple U-bend specimen, use	of Haimart Drucki call atraccad
specimens, application of the Allison bend test for detection emittance measurements, and analyses for the hydrogen and titanium, and for the chlorine content of manufacturing aid a specimentest parameter combination a rating of "compatible assigned."	of surface embrittlement. oxygen content of exposed materials. For each material-
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PREFACE

This is one of a series of final reports on Titanium Materials Technology submitted in fulfillment of Task 1-A of Department of Transportation contract DOT-FA-SS-71-12, dated 30 June 1971. The report was prepared by the Materials Technology organization of The Boeing Company, Commercial Airplane Group, Seattle. Washington.

This document is volume II of a two-volume series. Volume I covers flyaway materials, volume II covers manufacturing aid materials.

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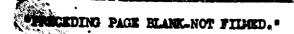
1.0 INTRODUCTION

This document records experience acquired by The Boeing Company concerning the compatibility of titanium alloys with aircraft manufacturing materials. The experience was gained during the Supersonic Transport (SST) program. Some of the data had been reported previously in an internally released Boeing Document (ref. 1).

This document is Volume 2 of a two volume series and covers manufacturing-aid materials, which are described as those substances such as drilling fluids, cleaners, etchants, or other materials which contact titanium alloys during component fabrication. In addition, a few substances (food stuffs, chemicals, etc.) which might accidentally contact titanium during shop handling were included. Volume 1 of this document (ref. 2) covers flyaway materials, such as coatings, sealants, oils, fuels, etc. which could contact the titanium alloys during normal aircraft operation.

Several of the tests used to determine the data recorded in this volume are described in detail in Volume 1, which also contains a short background discussion of the problem of stress corrosion cracking of titanium.

A rating of compatible (A) or incompatible (X) has been assigned for each material-alloy-exposure condition tested. An A rating does not constitute a recommendation for selecting a manufacturing aid material; obviously this requires an engineering judgement based upon all the material properties and the requirements of the proposed use. An X rating does not rule out using a material in contact with titanium in applications where it would encounter environmental stresses significantly lower than those used in the test.



2.0 BACKGROUND

Limited attention was paid to the stress corrosion behavior of titanium or its alloys prior to 1955, since it was not believed that titanium was prone to the problems that were involved with other alloy systems (ref. 3). In late 1955 it was discovered that a salty finger-print was the source of a stress-corrosion failure of a Ti-6A1-4V specimen during a 700°F creep test. This stress-corrosion effect of hot salt on titanium alloys was confirmed by other investigators (refs. 4 and 5).

In early 1966 B.F. Brown of the Naval Research Laboratory reported (ref. 6) that a titanium alloy (8 A1-1Mo-1V) was susceptible to stress-corrosion at room temperature. The Apollo fuel tank failure of December 1966 (ref. 7) during pressure testing using methyl alcohol served to intensity the study of environmental effects on titanium and titanium alloys.

Hydrogen embrittlement is another source of titanium alloy degradation. It can result from improper wet processing (typically nitric-hydrofluoric acid solutions) and from gaseous water during heat treating (ref. 8), and can result in seriously reduced fatigue resistance of parts. A related problem is titanium oxidation at high temperatures, producing a thick brittle oxide coating, which must be removed by a pickling process.

During manufacture of airplane parts, titanium will contact a wide variety of substances, collectively referred to as manufacturing aid materials. Each of these represents a potential source of stress-corrosion cracking and hydrogen embrittlement. For this reason Boeing began to accumulate test data on the compatibility of manufacturing aid materials with titanium alloys early in the SST program. The present document summarizes this test program.

Deterioration may occur not only during the manufacturing process involving a manufacturing aid, but also at a later time during a subsequent high temperature operation (e.g., stress-relief) if residual material is present. For this reason data on stress-corrosion is included at temperatures far above actual use values.

The data presented herein was acquired over a five year period. During this time an evolution of test methods occurred, and emphasis shifted from Ti-8Al-1Mo-1V to Ti-6Al-4V, the structural alloy eventually selected for SST use. Some data were also obtained on other titanium alloys, including Ti-4Al-3Mo-1V, Ti-5A-2.5 Sn, and CP titanium. Simultaneously, manufacturing methods were evolving. Some of the earlier data may not be as informative as some of that obtained later in the program, and direct comparison of materials tested early and late may not be possible. Nevertheless, the data is sufficient to guide the intelligent selection of manufacturing aid materials in order to avoid incompatibility problems.

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3.0 TEST METHODS

Brief descriptions are given in this section of the tests used in obtaining the compatibility data listed in section 5, including methods for detecting the occurance of embrittlement, corrosion, or stress corrosion, and the chemical analytical techniques used to identify test materials and to measure exposure produced changes in titanium alloy composition. Detailed laboratory directions for several of these are given in volume 1.

Early in the program the modified Allison bend test was the primary method used to detect attack on fitanium. This test can indicate the formation of a brittle surface layer due to either hydrogen embrittlement or oxidation. Since the specimen is exposed unstressed it will not detect susceptibility to stress corrosion cracking (SCC). A regular sequence of tests was employed including:

- Exposure of specimen coupons to the test material at the pertinent time-temperature conditions.
- Microscopic examination of the specimen surface for corrosion or etching.
- Allison bend testing to detect formation of a brittle surface layer.
- Gas analysis of broken Allison bend specimens to detect increased hydrogen content.

During the latter portion of the test program, when the importance of stress-corrosion cracking as a titanium degradation mechanism was more fully appreciated, emphasis shifted to the use of the U-bend test as the primary method of detecting incompatibility. This test, described in detail in Vol. 1, although not quantitative in nature, is inexpensive, relatively rapid, sensitive, and is admirably suited for a screening program in which large numbers of materials are to be tested.

In studying heat treat coatings the emittanc of the surface following exposure was measured to detect the presence of an oxide coat and to obtain some indication of its thickness. An additional thickness measurement water made by removing various thicknesses by chemical milling, according to BAC 5842, antil the emittance value of the milled specimen had dropped back to that typical of the bare metal.

3.1 SPECIMEN PREPARATION AND EXPOSURE

The majority of the data in this report deals with the SST structural titanium alloy containing 6 percent aluminum and 4 percent vanadium (Ti-6-4). Data on two other alloys is included: Ti-8-1-1, containing 8 percent aluminum, 1 percent molybdenum and 1 percent vanadium; and Ti-4-3-1, containing 4 percent aluminum, 3 percent molybdenum, and 1 percent vanadium. A few tests on pure titanium and on titanium containing 5 percent aluminum and 2.5 percent tin are also included. Test specimens were fabricated from flat stock 0.040 to 0.050-inches thick. Specimens were cleaned using a standard Boeing process (BAC 5753,

Method 2), which is described in volume 1 of this report (ref. 2). Unless otherwise indicated, specimen sizes depended upon the particular test to be run, as described below.

3.1.1 Continuous Exposure Test (Flat Sheet Exposure)

For yous ex sure tests candidate materials (lubricants, tapes, etc.) were brushed or otherw field to 5 x 6.00 inch sheets, and exposed at varying temperatures depending upon the 1. If m. sail being cyaluated and the usage conditions being simulated.

3.1.2 Intermittent Exposure Test (Drlp Corrosion)

An intermittent exposure test was originally designed to evaluate the effects of hot Skydrol on titanium and other metals. This test procedure was also used for the evaluation of the effects of machining lubricants and other fluid materials on titanium.

A 1.75×6 -inch specimen of titanium was suspended at an angle of $20\text{-}30^\circ$ in a heated pot. A burette containing the candidate fluid was positioned and adjusted so that a steady drip of material impinged on the top section of the specimen. Vacuum scaling tape was used along the edges of the specimen to maintain the flow of fluid the entire length without going over the edges and coating the under surface. Time at temperature varied depending upon the effects noted in the preliminary hours but generally testing was overnight or 16 to 17 hours.

Depending upon the size of the pot, a series of specimens may be used with a separate burette for each specimen. A control specimen was exposed at the same temperature but without contact with candidate material.

3.2 HEIMERL-BRASKI SELF-STRESSED SPECIMENS

This method for detecting stress corrosion employs two end-joined titanium coupons stressed to some predetermined level during material exposure by insertion of a cylindrical wedge. Specimen dimensions, specimen fabrication, application of test materials, environmental exposure, and compression testing following exposure as used in this program are described in detail in volume 1 of this document (ref. 2), together with stress-level calculations for this specimen configuration. The original description of the method by Heimerl and Braski is given in references 5 and 9.

3.3 U-BEND AND RESIN KETTLE TEST

The U-bend test is described in detail in volume 1 of this document (ref. 2). The single U-bend (S-U) specimen is a coupon bent into the form of the letter U, with dimensions such that the metal is stressed near to or above its yield point. Specimen configuration is shown in figure 1. A test material is applied to the convex side of the bend, and the specimen is exposed to the test environment. The metal surface is then cleaned and examined for stress cracking and corrosion. A rating of A or X is assigned to the alloy-material-environment combination by applying specified criteria, as listed in table 1.

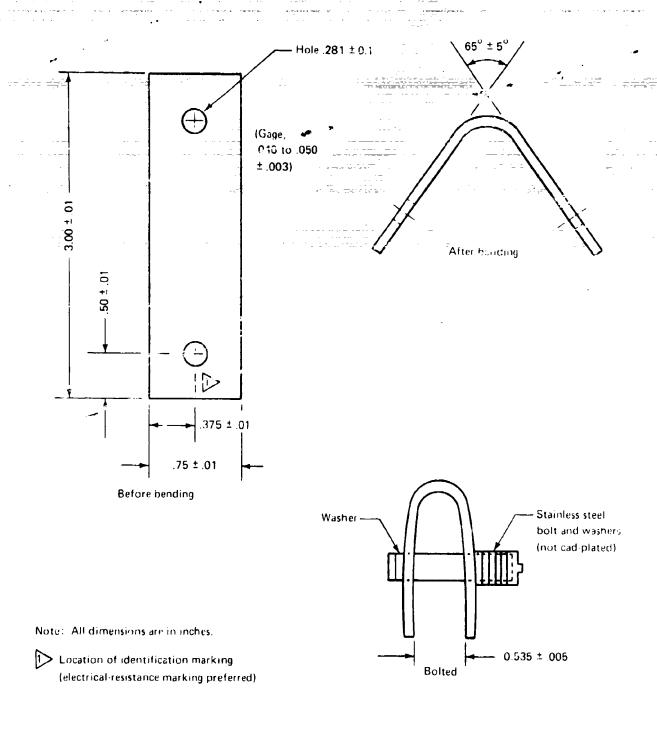


FIGURE 1.-U-BEND SPECIMEN

Modifications of this test described in volume I are the double U-bend test (D-U) in which the test material is trapped between two nested U-bends to simulate faying surface exposure, and the indented U-bend (I-U) in which small indentations along the center line of the bend are used to increase test sensitivity.

3.3.1 U-bend Test (Ambient)

This test was first used to evaluate the stress-corrosion effects of methanol on titanium but was subsequently extended to include other liquid materials. The test is simple a U-bend specimen is immersed in the material which is at room temperature. Exposure times to failure have been noted from a few minutes to 75 hours depending upon the titanium alloy, surface conditions, and the liquid environment.

The standard test employed specimens chemically cleaned per Method 2, BAC 5753. Variations in surface condition were found to be important and, therefore, care was exercised in noting exact surface history. All specimens were used un-notched, unless otherwise noted.

The importance of the stress level in determining sac epablility to corrosion is shown in figure 2. Small differences in the included angle between the legs of the specimen do not affect the time to fracture appreciably. This is because the apex of the bend is at or above the yield stress of the alloy for all small angles. However as the included angles rises above about 15° the applied stress decreases rapidly, and the susceptibility to corrosion, as shown by time to rupture, increases correspondingly. In considering this data it must be recognized that methanol, even at room temperature, represents an extremely severe environment for titanium alloys.

3.3.2 U-bend Test (Elevated Temperature)

The cleaned and stressed U-Bend specimens were dipped into or coated with candidate material. The specimens were drained with the open end up so that material would concentrate on the area of maximum stress. After overnight ambient drying the specimens were placed in a pre-heated, electrically fired oven typically for four hours at 1000°F. After exposure the specimens were descaled per BAC 5753, Method 2. Immersion in Turco 4316 Scale Conditioner or equivalent for 30 minutes followed by a four-minute immersion in the HNO₃-HF-pickle removed all heat-treat scale. The descaled specimens were examined microscopically for evidences of surface etch or stress corrosion cracking. Specimens were sectioned and micrographs taken as required. All specimens were used un-notched, unless otherwise noted.

3.3.3 Resin Kettle Test

The term "Resin Kettle Test" was applied in this program to a test in which a specimen of the U-bend type was exposed to vapors evolved from a test fluid. Although the exposure vessel was not sealed, in order to allow venting, the openings were small, and a vapor atmosphere was maintained around the specimen by gravity effects for the test duration. This allowed detection of stress-corrosion produced by decomposition products of the vapor on the titanium alloy surface.

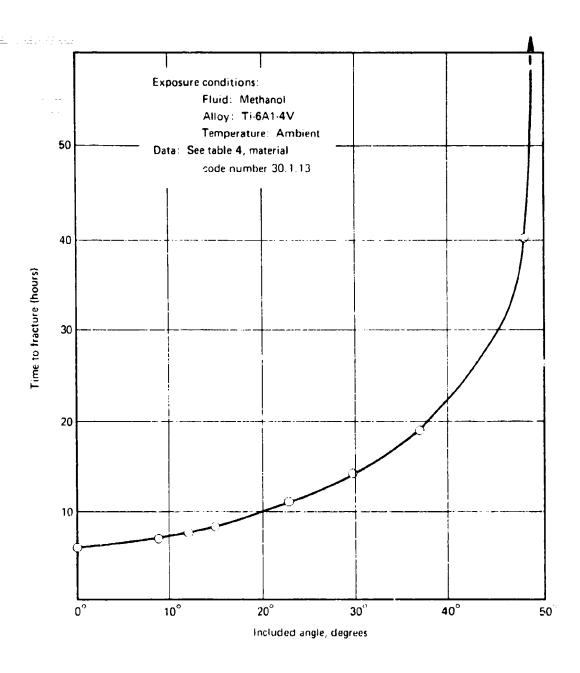


FIGURE 2. -TIME TO FRACTURE OF U-BEND SPECIMENS RELATED TO THE INCLUDED ANGLE BETWEEN SPECIMEN SIDES

The reaction chamber used for testing was a resin kettle of 4000 ml capacity. A typical test was as follows:

- The glass resin-kettle apparatus was chemically cleaned and baked at 350°F for 3 to 6 hours.
- Two Ti 6A1-4V and Ti 8A1-1Mo 1V alloy U-bend specimens (alkaline and HNO₃-HF pickle cleaned before restraint) were placed in the bottom of the kettle with the stressed surface upward.
- The lid of the kettle was secured with two openings left open for fume exhaust.
- The material in question, when a liquid, was poured into the kettle through an opening in the lid, whereupon the whole assembly was placed in a preheated explosion-proof, vented oven. A volume of 35-38 ml was used.
- Time at temperature varied with desired simulated exposure, but was generally only two hours at 850°F. (Time was found not to be a significant factor.)
- After exposure, the specimens were inspected for cracks. When required, wet penetrant inspection and microscopic examination were used.

3.4 MODIFIED ALLISON BEND TEST

The Allison bend test was originally developed to study the toughness and crack propagation properties of steel alloys (ref. 10). Work at Bocing had shown that results of this test were strongly affected by the surface condition of the test coupons, as well as by the bulk properties of the alloy. Because of this observation, this test was chosen to study surface embrittlement of titanium alloys produced by exposure to manufacturing aid materials.

In the test thin gage specimens are bent to fracture over a small mandrel, using a fixture as shown in figure 3. The load deflection curve is recorded during this process. Figure 4 illustrates load-deflection curves typical of a non-embrittled control specimen and of a specimen exhibiting severe surface embrittlement. The two curves are approximately the same from the origin to point A (at the maximum stress) where the convex surface begins to yield. Curve A-E traces the continued yielding of the control coupon until it fractures, allowing the stress to go to zero along E-D. If the alloy is sufficiently tough, fracture may occur in steps leading to curves such as the dashed line E-D'. The area A-B-D-E-A (or A-B-D'-E-A) is measured using a planimeter. Expressed in inch-pounds it is called the bend-energy of the control.

In an embrittled specimen small surface cracks develop at the yield point A. These act as stress risers, so that the curve drops off to the axis along A-C (or A-C') with fracture occurring at a lesser head travel distance. The bend energy (area of A-B-C-A or A-B-C'-A) is measured and compared to that of the control. A decrease of 15-percent or greater in the bend energy, relative to the control is considered significant evidence of surface embrittlement. This method of data interpretation differs from that originally suggested by Hanik in which the differences between the maximum stress and the stress at which fracture is initiated (i.e. stress at A stress at E for the control) are compared for various specimens.

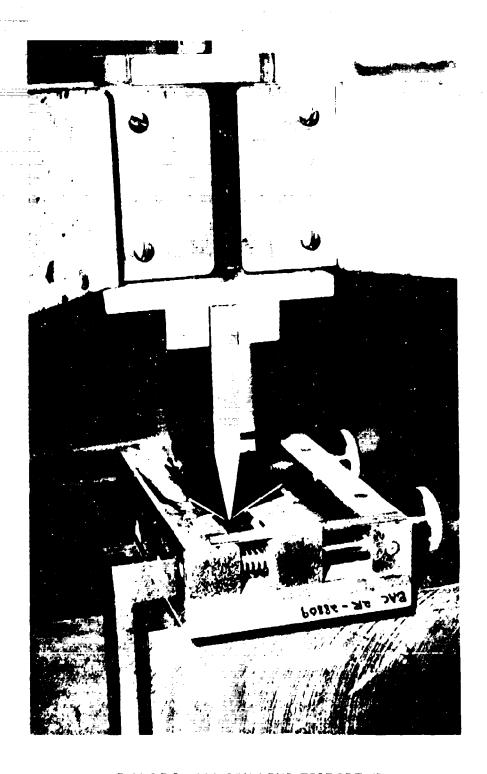


FIGURE 3. -ALLISON BEND TEST SET UP

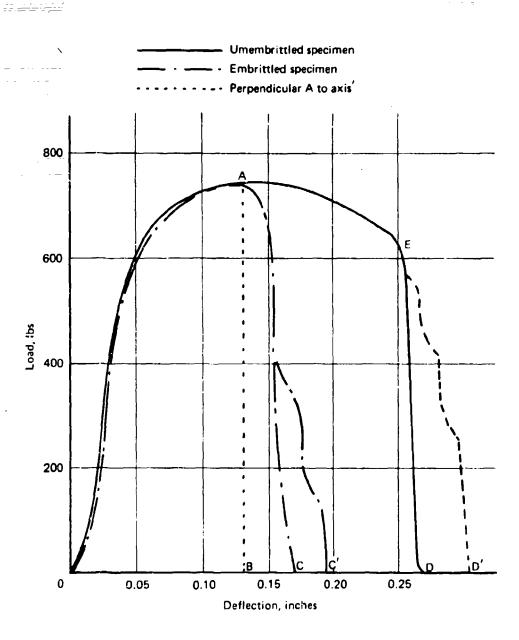


FIGURE 4.--LOAD DEFLECTION CURVE, ALLISON BEND TEST

A standard specimen is fabricated from 0.040 or 0.050-inch sheet stock, and is 1.75×2.00 inches, with the 2-inch dimension along the rolling direction. Specimens are cleaned according to BAC 5753, Method 2, and exposed as described in 3.1.1 above. Fracture is induced in the long or rolling direction, using a 0.034-inch radius mandrel. The channel width of the jig is set at 14-times the specimen thickness, and the head travel speed is 0.25 inches per minute.

3.5 EMITTANCE TEST

This test is based on the correlation between the degree of oxidation of a titanium surface, and the surface emittance. It was used to evaluate the ability of heat treat protective coatings to protect titanium surfaces from oxidation. Emittance is defined as the ratio of energy radiated per unit time from a unit surface area from a body to the energy radiated from a perfect black body at the same temperature.

The emittance of materials is a surface phenomenon. A metal surface in its natural state exhibits a certain emittance that is characteristic of that particular metal. The same metal covered with another layer, such as an oxide film or a synthetic coating, possesses an emittance property which is generally significantly different from that of the natural metal surface. Very thin films are very nearly transparent to radiant energy and their emittance properties are greatly influenced by the metal substrate. When the coating on a metal surface is sufficiently thick and opaque, the emittance of the composite is that of the coating itself.

The following typical data illustrates how emittance measurements can furnish information about the state of titanium surfaces.

Total Emittance

<u>Titanium Alloy</u>	<u>Values</u>
Ti 6AI-4V	
Theoretical (clean & smooth)	0.143
Chemically Cleaned	0.15
As-Received	0.19
400 ^o F Oxidized	0.21
Anodized Titanium	ti.
Gold	0.14
Blue Gray	0.17
Gray Black	0.32
Gray	0.38

In studying heat treat coatings the emittance of the surface following exposure was measured to detect the presence of an oxide coating, and to obtain some indication of its thickness. An additional thickness measurement was then made by removing various thicknesses by chemical milling, according to BAC 5842, until the emittance value of the milled specimen had dropped back to that typical of the bare metal.

The instrument used for this study was the Lion Research Corporation Model 25B Emissometer. Briefly this instrument consists of a transducer head, a transducer output signal indicator, a thermo-electric cooler supply, a water circulater, and a vacuum system. These components are mounted in a console which is equipped with rubber wheels for portability.

In operation, the head (equipped with an eight-foot extension cable) is placed against the surface to be tested. Radiation from the surface enters the head and is absorbed by the blackened receiver disc of a thermopile. The receiver disc re-emits to its cooled surroundings causing the receiver disc to assume some temperature between the sample surface temperature and the temperature of the cold platten. The temperature of the thermopile receiver is dependent on the amount of radiation received which is related to both the emittance and temperature of the test surface. However, if the test surface temperature and the temperature of the cooled surfaces around the thermopile are kept constant, the thermopile output voltage is dependent only on the emittance of the test surface. The values of emittances are directly read off the meter scale mounted on the instrument console.

For each set of measurements the emissometer is calibrated against emittance standards, kept at the same temperature as the test specimens. This corrects for temperature changes from day to day, as well as for instrument response charges. The calibration standards are maintained by the Boeing Standards Group.

3.6 CHEMICAL TESTS

The tabulated data contains results of a number of chemical tests. These were carried out for two purposes: first to obtain information about the chemical composition of commercial products of unknown character, and second to determine the amounts (percent) of certain constituents, halogen atoms and ions, known to produce stress corrosion cracking of titanium.

3.6.1 Infrared Spectroscopy

Standard techniques were employed to measure the infrared spectra of interest. These included use of KBr pellets, and liquid film samples. Spectra were measured from 2.5-15 microns.

3.6.2 Chloride Analysis

Several methods of chloride analysis were used. These are methods used in the Boeing Quality Control Laboratories. They are listed below, and are described in detail in Appendix 1.

- Bosler method For chlorides in machining lubricants
- Radiotracer silver method For soluble trace chlorides
- Beilstein test For qualitative detection only

- Parr bomb method For chlorides in dye penetrants and other combustible samples
- Sodium carbonate fusion For halides in non-combustible samples

3.6.3 Dissolved Gas Analysis

Titanium has an affinity for hydrogen, and hydrogen absorption can lead to embrittlement. Two closely related analytical methods were used to measure the hydrogen content of titanium specimens exposed to various test materials, in order to detect hydrogen intake. This data, in conjunction with the results of the modified Allison bend tests, permitted an evaluation of possible incompatibility due to hydrogen embrittlement.

3.6.3.1 Vacuum Fusion

Vacuum Fusion is a technique which is capable of determining the hydrogen, oxygen, and nitrogen contents of most metals. The analyzer is essentially an evacuated system containing a molten platinum bath which is held between 1800 degrees and 2100 degrees C. A carbon element and crucible are used to attain these temperatures. Weighed samples, when placed into the platinum bath, release their gaseous contents. Oxygen is converted to carbon monoxide.

The gas is immediately pumped on to the analytical section and a McLeod gauge reading taken. Hydrogen is oxidized to water and carbon monoxide to carbon dioxide by passage over hot copper oxide. After removing the hydrogen by conversion to water and cold trapping, another McLeod gauge reading is taken. A final reading is taken after the carbon dioxide is removed with Ascarite. The last reading is assumed to be nitrogen. By comparing these readings to those of a standard or to the gas law (analytical volumes already known), the quantities of gas per weight of sample may be calculated.

The sample size used in vacuum fusion is between 0.03 and 0.10 grams with an average of .06 grams. This relatively small sample size requires extreme care in surface cleaning and sample cutting. Volatile solvents, such as benzene, xylene, and acetone remove any organic contaminant, and filing removes any surface crust. Sanding should be avoided due to its tendency to produce high results.

Once the specimen is cleaned, a sample is cut with a chisel (the preferred method) or with a hacksaw, whose blade has been cleaned in benzene. Lathing and milling also seem to be acceptable; however, they are dependent on the care taken by shop personnel.

There are two means for standardization of vacuum fusion: (1) Theoretical calculations using the ideal gas law and (2) Comparison of sample data with data on standards of known gaseous content, processed identically.

 Several attempts have been made to standardize vacuum fusion with the ideal gas law. Success has been achieved however, only under ideal conditions for hydrogen and oxygen. One major task in establishing a correlation between calculated results using the gas law and empirical results based on known standards was the estimation of the gas collection volumes. These can now be accurately determined with a dosing stopcock and argon gas. A fresh platinum bath (one with negligible titanium contamination) is also required to prevent low results, indicating incomplete release of the gases from a contaminated bath.

2. The National Bureau of Standards has not produced an oxygen standard for titanium, although it has several pure titanium hydrogen standards. Therefore, it was advisable to standardize one of the NBS hydrogen standards for oxygen as a reference to use throughout this program. NBS 354 was chosen because of its reproducible oxygen readings and was established at 4500 ppm by many carefully conducted gas law determinations.

3.6.3.2 Hot Extraction Method

The accuracy attainable for hydrogen using the vacuum fusion was marginal for the purposes of this program. The hot extraction method utilizes the same test apparatus; however the test sample (approximately 100 mg) is heated to a lower temperature (1400C) in a carbon crucible. At this temperature, below the melting point of titanium, only hydrogen is evolved from the sample, which still retains its oxygen and nitrogen. The hydrogen can thus be determind directly from a single McLeod gage reading. Testing during this program indicated this to be a more precise analytical method.

3.7 METALLOGRAPHIC TESTS

Standard metallographic techniques were used for the evaluation of stress-corrosion, intergranular attack, etc., as required.

3.8 MACHINING

Standard machine shop equipment was used to perform drilling, grinding, and milling on titanium specimens, as required in the evaluation of drilling or machining fluids.

4.0 INTERPRETATION OF TEST RESULTS

Compatibility ratings of A (compatible) or X (incompatible) were derived from the test data for each material listed in table 4, section 5. The criteria used in deriving these ratings are described below. For some materials the rating was derived from the results of several tests. In the tabulated data this is indicated by placing the compatibility rating on the same line as the name of the material, and listing the various test and results below. For other materials individual compatibility ratings were derived for each time-temperature-alloy exposure condition. This is indicated by placing the rating on the same line as the test decription.

In applying the compatibility ratings caution is required. An A rating should not be considered to imply compatibility under exposure conditions more severe than the maximum tested. A material compatible at several hundred degrees, its use temperature, may produce corrosion if not completely removed before a subsequent heat treating operation at a much higher temperature.

The most useful test for detecting incompatibility proved to be the U-bend test in its several modifications. In extreme cases incompatibility manifested itself by actual specimen fracture during exposure. In less severe cases varying degrees of surface attack could be detected by microscopic examination of the stressed surface following descaling. Table 1 lists five levels of severity of attack which could be distinguished by careful examination, and relates these to compatibility ratings of A or X. (Photographs illustrating SCC levels 2-5 are included in Vol. 1 of this report (ref. 2)).

TABLE 1.-LEVELS OF SEVERITY OF STRESS CORROSION CRACKING (SCC)

Level	Effect Observed	Compatibility Rating
	No Surface Effect	Α
0	Staining, Shadowing, or Frosting	Α
0	Etching or pitting smoother than 32 RHR	Α
0	Etching or pitting rougher than 32 RHR	X
1	Slight SCC, a few short cracks	X
2	Moderate SCC, longer cracks, wider distribution than level 1	X
3	Widespread, general SCC	X
4	Massive general SCC	X
5	Complete failure, fracture	X

The criteria used to assign an A or X rating based on the Heimerl-Braski test are discussed in detail in reference 2. Briefly a rating of X is given if:

• One or more material exposure specimens fractured during exposure

- None of the material exposure specimens exhibited a deflection distance equal to
 or greater than the smallest deflection distance exhibited among the applicable
 control specimens.
- The statistical t value was greater than 3.7.

The statistical t value, for quadruplicate specimens, is defined by the equation

$$t = \frac{3.46 (\bar{X}_c - \bar{X}_m)}{(\Sigma (X_c - \bar{X}_c)^2 + \Sigma (X_m - X_m)^2)^{\frac{1}{2}}}$$

c = Control (uncoated)

m = Material coated

X = Observed head travel

 \bar{X} = Average value for four specimens.

All materials not classified as X by one of the above are given an A rating.

Assignment of A or X ratings based upon chloride analysis proved to be difficult. Initially it had been planned to assign an X rating to materials containing more than 200 ppm of chlorine, since chloride ion is a major cause of titanium stress-corrosion cracking. Test results, however, especially on dye penetrant materials, showed poor correlation between chlorine levels and SCC. For this reason X ratings were not assigned based upon chlorine content alone without confirming evidence of surface attack. However it would be wise to be cautious in using materials containing high chlorine contents. Materials containing chlorinated solvents were rated X, even at temperatures where the pure solvents might be compatible, due to the possibility that chlorine could be retained in contact with the titanium to higher temperatures.

The results of the Allison bend testing are described in section 6. A bond energy difference of greater than 15 percent between control and test specimens probably indicates some surface change. The test detected oxide films formed during high temperature exposure, but no instances were found where large enough differences between exposed and control specimens in the same environment were found to indicate material incompatibility.

5.0 RESULTS OF COMPATIBILITY TESTING

The data obtained by application of the previously described test methods is tabulated in this section. An explanation of the coding used to identify test materials and parameters is given, together with an alphabetical index of the materials tested.

5.1 MATERIAL NUMBER CODE

The material number code consists of three index numbers. The materials tested for compatibility with titanium alloys have been divided into thirteen major classes, such as cleaner or scale conditioner, indicated by the first index number, beginning with 18. (Classes 1 - 17 are the flyaway materials covered in Vol. 1 of this report.) The second index number identifies specific subclasses such as protective coating, strippable. The third index number identifies one specific material within a class. Test data tabulated in the following section of this report is arranged in numerical order according to this code. Table 2 contains a key to the classes and subclasses.

5.2 ALPHABETICAL INDEX OF MATERIALS TESTED

Table 3 lists the materials tested for titanium compatibility in alphabetical order, with references to the location of the data in table 4. Vendors are identified in the tabulated data, and in table 3.

TABLE 2.—KEY TO CLASSES AND SUBCLASSES: MANUFACTURING AID MATERIALS

NUMBER	CLASS	SUBNUMBER	SUBCLASS
18	Cleaner		
19	Food		
20	Heat Treat Aid		
21	Machining Lubricant/Coolant	}	
22	Marking, Electrochemical		
		1	Cleaner
		2	Electrolyte
23	Marking, Temporary		
		1	Heat Treat Indicator
		2	Ink, Temporary Marking
		3	Pencil
24	Metal		
25	Paint		
		1	Leak Detection
		2	Protective
26	Penetrant Inspection Fluid		
1		1	Developer
		2	Emulsifier/Remover
		3	Penetrant
27	Protective Coating		
		1	Heat Treat
	Į	2	Spinning Compound
		3	Strippable (Temporary)
28	Salt		
29	Scale Conditioner		
	1	1	Aqueous
		2	Molton Salt
30	Solvent		
31	Stabilizer, Machining	ļ	
32	Stripper, Paint		
33	Tape, Temporary		
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TABLE 3.-ALPHABETICAL INDEX OF MATERIALS TESTED

Material Name or Designation	Material Type	Vendor	Number Code
A-498	Protective Coating	Glidden-Metlseel	27.1.11
Acetic Acid	Solvent		30.1.1
Ardrox 985-P2	Penetrant Inspection Fluid	Ardrox Australia PTY, Ltd	26.3.1
Ardrox 9D-3	Penetrant Inspection Fluid	Ardrox Australia PTY, Ltd	26.1.1
Ardrox 9PR-4	Penetrant Inspection Fluid	Ardrox Australia PTY, Ltd	26.2.1
Ammonium Hydroxide	Solvent	·	30.1.2
APC	Marking, Electrochemical	Marking Methods, Inc.	22.1.5
Aquamatic Universal Coolant	Machining Lubricant/Coolant	Molecular Products, Inc.	21.1.12
8-10	Marking, Electrochemical	Monode, Inc.	22.2.11
B-410	Machining Lubricant/Coolant	Texaco, Inc.	21.1.43
Barium Hydroxide	Machining Lubricant/Coolant	,	21.1.1
Barkerwax 3BE	Stabilizer, Machining	Barker Enterprises	31.1.7
Blaisgell 551 Blue	Marking, Temporary	Blaisdell Co.	23.3.1
Boeing, Proprietary	Scale Conditioner	Boeing Co.	29.2.1
Bon Ami	Cleaner	3	18.1.2
Boric Acid	Protective Coating		27.1.4
Boric Acid GE SR350 Silicone	Protective Coating		27.1.5
Boron Nitride	Heat Treat Aid		20.1.1
Brandt 8224	Scale Conditioner	Brandt Chemical Co.	29.1.1
Butyl Alcohol	Solvent		30.1.3
Butyl Carbitol	Solvent	Union Carbide Corp.	30.1.4
Butyl Cellosolve	Solvent	Union Carbide Corp.	30.1.5
C-250	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.13
C-300	Protective Coating	Fel-Pro Corp.	27.1.10
Carbon Tetrachloride	Solvent		30.1.6
Carus 303	Scale Conditioner	Carus Chem, Co.	29.2.2
Cerrolow 140	Stabilizer, Machining	Cerro de Pasco Corp.	31.1.8
Cerrolow 174	Stabilizer, Machining	Cerro de Pasco Corp.	31.1.9
Cool·Tool	Machining Lubricant/Coolant	Munroe Chem. Co.	21.1.25
Coollube No. 10	Machining Lubricant/Coolant	Pacific Chem, Co.	21.1.26
Covington Cote	Protective Coating	Timet Corp.	27.1.29
Crystal Cut 322	Machining Lubricant/Coolant	Hangsterfer's Labs, Inc.	21.1.5
Cut Max 569	Machining Lubricant/Coolant	E.F. Houghton & Co.	21.1.8
D-70	Penetrant Inspection Fluid	Met-L-Chek Co.	26.1.11
D-90	Penetrant Inspection Fluid	Sherwin, Inc.	26.1.19
D-100	Penetrant Inspection Fluid	Sherwin, Inc.	26.1.20
D-110	Penetrant Inspection Fluid	Sherwin, Inc.	26.1.21
D-113A	Penetrant Inspection Fluid	Sherwin, Inc.	26.1.22
D-492B	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.1.12
D-492C	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.1.13
D-493	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.1.14

TABLE 3.-ALPHABETICAL INDEX OF MATERIALS TESTED (continued)

Material Name or Designation	Material Type	Vendor	Number Code
D-493A	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.1.15
D-495A	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.1.16
D-499B	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.1.17
D-499C	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.1.18
DD-2	Penetrant Inspection Fluid	Fluro-Chek Co.	26.1.2
DD 535	Penetrant Inspection Fluid	Sperry Div., Automated Industry, Inc.	26.1.24
Del-Chem 19AC	Stripper, Paint	•	31.2
Delta 31	Protective Coating	Acheson Colloids, Inc.	27.1.1
DGS	Scale Conditioner	Kolene Co.	29.2.4
Dickson 1107 NH	Marking, Temporary	Dickson Ink Co.	23.2.1
Dickson 1307	Marking, Temporary	Dickson ink Co.	23.2.2
Dickson 2101	Marking, Temporary	Dickson Ink Co.	23.2.3
Dickson 2307	Marking, Temporary	Dickson Ink Co.	23.2.4
Dickson 5507	Marking, Temporary	Dickson Ink Co.	23.2.5
DTE No. 24	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.14
DTE No. 26	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.15
DW 530	Penetrant Inspection Fluid	Sperry Div., Automated Industries, Inc.	26.1.25
Dy-Chek, Developer	Penetrant Inspection Fluid	Turco Products, Inc.	26.1.29
Dy-Chek, Emuisifier	Penetrant Inspection Fluid	Turco Products, Inc.	26.2.15
Dy-Chek, Penetrant	Penetrant Inspection Fluid	Turco Products, Inc.	26.3.37
E-56	Penetrant Inspection Fluid	Met-L-Chek Co.	26.2.8
E-59	Penetrant Inspection Fluid	Met-L-Chek Co.	26.2.9
E-142	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.2.10
E-153	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.2.11
E-157A	Penetrant Inspection Fluid	Shannon Luminous Materials Co.	26.2.12
£	Penetrant Inspection Fluid	Turco Products, Inc.	26.2.6
Eagle Oil No. 10	Machining Lubricant/Coolant	Richfield Oil Co.	21.1.38
Electrolyte 359L	Marking, Electrochemical	Electro Chem Etch Co.	22.2.1
Electrolyte 1	Marking, Electochemical	Electromark Corp.	22.2.3
Electrolyte 2	Marking, Electrochemical	Electromark Corp.	22.2.4
Electrolyte 4	Marking, Electrochemical	Electromark Corp.	22.2.5
Electrolyte 59-L	Marking, Electrochemical	Electromark Corp.	22.2.6
Electrolyte 353	Marking, Electrochemical	Electromark Corp.	22.2.7
Electromark 2-1	Marking, Electrochemical	Electromark Corp.	22.1.2
Electromark 2-2	Marking, Electrochemical	Electromark Corp.	22.1.3
ER 82	Penetrant Inspection Fluid	Sherwin, Inc.	26.2.13
ES&T	Marking, Electrochemical	Electro Chem Etch Co.	22.2.2
Ethanol	Solvent		30.1.7

TABLE 3.-ALPHABETICAL INDEX OF MATERIALS TESTED (continued)

Material Name or Designation	Material Type	Vendor	Number Code
Ethylene Glycol	Solvent		30.1.8
Epoxy Enamel	Paint	BMS 10-11	25.2.2
Epoxy Primer	Paint	BMS 10-11	25.2.1
F-10	Marking, Electrochemical	Monode, Inc.	22.2.12
F-20	Marking, Electrochemical	Monode, Inc.	22.2.13
FD-33	Penetrant Inspection Fluid	Testing Systems, Inc.	26.1.26
FH-420	Protective Coating	Acheson Colloids Co.	27.1.2
Fiber-Frax	Heat Treating Aid		20.1.2
FL-50	Penetrant Inspection Fluid	Testing Systems, Inc.	26.3.36
Fluoro Finder (Dry)	Penetrant Inspection Fluid	Testing Systems, Inc.	26.1.27
Fluoro Finder (Wet)	Penetrant Inspection Fluid	Testing Systems, Inc.	26.1.28
FP-22	Penetrant Inspection Fluid	Sherwin, Inc.	26.3.31
FP-90	Penetrant Inspection Fluid	Met-L-Chek	26.3.13
FP-91	Penetrant Inspection Fluid	Met·L·Chek	26.3.14
FP-92	Penetrant Inspection Fluid	Met-L-Chek	26.3.15
FP-93	Penetrant Inspection Fluid	Met-L-Chek	26.3.16
FP-95	Penetrant Inspection Fluid	Met-L-Chek	26.3.17
	1		26.3.35
FPE-505	Penetrant Inspection Fluid	Sperry Div., Automated Industries, Inc.	
Freon-Butyl Cellosolve	Machining Lubricant/Coolant	DuPont Co.	21.1.3
Freon 1301	Solvent	DuPont Co.	30.1.9
Freon MF	Solvent	DuPont Co.	30.1.10
Freon PCA	Solvent	DuPont Co.	30.1.11
Gage Cote 6	Protective Coating	i	27.1.30
GMC 895	Cleaner	Greater Mountain Chem. Co.	18.1.3
Grind-Tex B-410	Machining Lubricant/Coolant	Texaco Co.	21.1.43
Habcool 318	Machining Lubricant/Coolant	H & B Petroleum Co.	21.1.4
HD 200	Protective Coating	Kerns Pacific Corp.	27.1.15
HD 205	Protective Coating	Kerns Pacific Corp.	27.1.16
HD 6330	Protective Coating	Kerns Pacific Corp.	27.1.17
Hexane	Solvent	•	30.1.12
HM-3	Penetrant Inspection Fluid	Sherwin, Inc.	26.3.32
HM-225	Penetrant Inspection Fluid	Sherwin, Inc.	26.3.33
HM-405	Penetrant Inspection Fluid	Sherwin, Inc.	26.3.34
Hocut 237	Machining Lubricant/Coolant	E.F. Houghton & Co.	21.1.9
Houghto Safe 620	Machining Lubricant/Coolant	E.F. Houghton & Co.	21.1.10
HS 8171 PS	Tape, Temporary	Richmond Corp.	33.1.6
Iodine, Toluene, and SAE 10	Machining Lubricant/Coolant	The state of the s	21.1.11
Ink	Marking, Temporary	Esterbrook Co.	23.2.9
Ink	Marking, Temporary	Sharpie Co.	23.2.12
Jarvie 40	Stabilizer, Machining	Jarvie Paint Co.	31.1.10
K-58-C	Stripper, Paint	Keelite Corp.	32.1.1
Kalcote XP	Protective Coating	California Coatings Co.	27.1.6
Kalgard FL (Kalube FL)	Protective Coating	California Coatings Co.	27.1.8

TABLE 3.-ALPHABETICAL INDEX OF MATERIALS TESTED (continued)

Material Name or Designation	Material Type	Vendor	Number Code
Kalgard FL-40	Protective Coating	California Coatings Co.	27.1.7
(Kalube FL-40)	Ĭ	•	1
Kelite 9	Paint	Kelite Chem. Co.	25.1.2
Kirksite	Metal		24.1.1
L-028	Penetrant Inspection Fluid	Sherwin, Inc.	26.1.23
Lead	Metal	·	24.1.2
LF-22	Protective Coating	Lubri-Film, Inc.	27.1.18
LF-32	Protective Coating	Lubri-Film, Inc.	27.1.19
LF-70	Protective Coating	Lubri-Film, Inc.	27.1.20
LF 75	Protective Coating	Lubri-Film, Inc.	27.1.21
LF-80	Protective Coating	Lubri-Film, Inc.	27.1.22
LF-500	Protective Coating	Lubri-Film, Inc.	27.1.23
LF-600	Protective Coating	Lubri-Film, Inc.	27.1.24
LF-700	Protective Coating	Lubri-Film, Inc.	27.1.25
M-3	Marking, Temporary	Marsh Co.	23.2.10
3M 850	Tape (Temporary)	3M Co.	33.1.1
Major Clean	Cleaner	···· • • • • • • • • • • • • • • • • •	18.1.4
Marking Methods 5	Marking, Electrochemical	Marking Methods, Inc	22.1.4
Mercury	Metal	waxing wethous, mo	24.1.3
Met 25	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.16
Methanol	Solvent	11.001.01.00.	30.1.13
Methyl Chloroform	Solvent		30.1.14
Methyl Ethyl Ketone	Solvent		30.1.15
Milk	Food		19.1.1
Missile Lube No. 1	Machining Lubricant/Coolant	Hangsterfer's Labs, Inc.	21.1.6
Missile Lube No. 5	Machining Lubricant/Coolant	Hangsterfer's Labs, Inc.	21.1.7
Mist No. 36	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.17
MSC No. 3	Marking, Electrochemical	Monode, Inc.	22.1.6
MSC No. 5	Marking, Electrochemical	Monode, Inc.	27.1.7
Mystic 5863	Tape, Temporary	Mystic Co.	33.1.3
Mystic 6110	Tape, Temporary	Mystic Co.	33.1.4
Nad Developer	Penetrant Inspection Fluid	Turco Products	26.1.30
Neutracleaner No. 1	Marking, Electrochemical	Electrochemical Etch Co.	22.1.1
Nevr-Dull	Cleaner	George Basch Co.	18.1.5
Orange Detector Paint	Paint	Fuller Co.	25.1.1
Organoceram 1-1043	Protective Coating	Organocerams, Inc.	27.3.2
Organoceram 1-1044	Protective Coating	Organocerams, Inc.	27.3.3
Organoceram 1-2020	Protective Coating	Organocerams, Inc.	27.3.4
P3F	Penetrant Inspection Fluid	NAA-Rockwell	26.3.18
P3F-3	Penetrant Inspection Fluid	NAA-Rockwell	26.3.19
P5F-1	Penetrant Inspection Fluid	NAA-Rockwell	26.3.21
P5F-2.5	Penetrant Inspection Fluid	NAA-Rockwell	26.3.21
P5F-3	Penetrant Inspection Fluid	NAA-Rockwell	26.3.22
P-40	Penetrant Inspection Fluid	Turco Products	26.3.38
P-133	Penetrant Inspection Fluid	Shannon Luminous	26.3.23
1 133	renetiant inspection i fuld	Products, Inc.	20.5,25

TABLE 3.—ALPHABETICAL INDEX OF MATERIALS TESTED (continued)

	Material Name or Designation	Material Type	Vendor	Number Code
	P-133A	Penetrant Inspection Fluid	Shannon Luminous Products, Inc.	26.3.24
, lustera	P-134	Penetrant Inspection Fluid	Shannon Luminous Products, Inc.	26.3.25
	P-134A	Penetrant Inspection Fluid	Shannon Luminous Products, Inc.	26.3.26
;	P-135	Penetrant Inspection Fluid	Shannon Luminous Products, Inc.	26.3.27
	P-148A	Penetrant Inspection Fluid	Shannon Luminous Products, Inc.	26.3.28
	P-149	Penetrant Inspection Fluid	Shannon Luminous Products, Inc.	26.3.29
	P-150	Penetrant Inspection Fluid	Shannon Luminous Products, Inc.	26.3.30
	Pannier 1001 Red	Marking, Temporary	Pannier Ink Co.	23.2.7
	Pasa-Jell 107C-7	Cleaner	Semco Sales & Service Corp.	18.1.1
	PE 520	Penetrant Inspection Fluid	Sperry Div., Automated Industries, Inc.	26.2.14
1	Pencil, Blue 551	Marking, Temporary	Blaisdell Co.	23,3.1
1	Perchlorethylene	Solvent	1	30.1.6
	Permacel 733	Tape, Temporary	Permacel Co.	33.1.5
	PF (Rigidax)	Stabilizer, Machining	M. Argüeso & Co., Inc.	31.1.2
	Polyurethane Enamel	Paint	BMS 10-60	25.2.3
	Pretreat	Protective Coating	Turco Products, Inc.	27.1.28
	n-Propyl Alcohol	Solvent		30,1.17
	iso-Propyl Alcohol	Solvent		30.1.18
,	Propylene Glycol	Solvent		30.1.19
	Project-A-Mark	Marking, Temporary	Marsh Co.	23.2.11
	RA-536	Protective Coating	Glidden-Metiseel	27.1.12
	RA-537	Protective Coating	Glidden-Metlseel	27.1.13
	RA-538	Protective Coating	Glidden-Metiseel	27.1.14
	Rapid Top	Machining Lubricant/Coolant	Relton Corp.	21.1.37
	Rigidax Blue F	Stabilizer, Machining	M. Argueso & Co., Inc.	31, 1.1
	Rigidax PF	Stabilizer, Machining	M. Argueso & Co., Inc.	31.1.2
	Rigidax W1 Blue	Stabilizer, Machining	M. Argüeso & Co., Inc.	31.1.3
	Rigidax WI Green	Stabilizer, Machining	M. Argueso & Co., Inc.	31.1.4
	Rigidax W1 NF	Stabilizer, Machining	M. Argueso & Co., Inc.	31.1.5
	Rigidax YF	Stabilizer, Machining	M. Argueso & Co., Inc.	31.1.6
:	SC-44	Marking, Electrochemical	Electromark Corp.	22.2.8
	SC-82	Marking, Electrochemical	Electromark Corp.	22.2.9
	SC-277-2	Protective Coating	Spraylat Corp.	27.3.5
	SC-1071	Protective Coating	Spraylat Corp.	27.3.6
	Silver Chloride	Salt		28.1.1
	Silver Iodide	Salt		28.1.2
	Silver Nitrate	Salt	N 6	28.1.3
	Snoop No. 3	Paint	Nupro Corp.	25.1.3
	Sodium Chloride	Salt		28.1.4

TABLE 3.-ALPHABETICAL INDEX OF MATERIALS TESTED (continued)

Material Name or Designation	Material Type	Vendor	Number Code
Sodium Hydroxide	Scale Conditioner		29.1.2
Sodium Nitrite	Machining Lubricant/Coolant		21.1.40
Solvac 2032	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.18
Solvac NP	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.19
Sultran 176-M	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.20
T-10 (Cimcool)	Machining Lubricant/Coolant	Cincinatti Milling Machine Company	21.1.2
T-10	Marking, Electrochemical	Marking Methods, Inc.	22.2.10
T-50	Protective Coating	Everlube Corp. of America	27.1.9
Tap Magic	Machining Lubricant/Coolant	Staco Corp.	21.1.41
Tapzol 410	Machining Lubricant/Coolant	Rust-Lick Corp.	21.1.39
Tempilstik 1150°F	Marking, Temporary	Tempilstik Co.	23.1.1
Tempilstik 1350°F	Marking, Temporary	Tempilstik Co.	23.2.2
TJ:73	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.21
Trichlorethylene	Solvent	mobile on Go.	30.1.20
Trikut DS	Machining Lubricant/Coolant	Pemaco, Inc.	21.1.27
Trikut DSL	Machining Lubricant/Coolant	Pemaco, Inc.	21.1.28
Trikut GO	Machining Lubricant/Coolant	Pernaco, Inc.	21.1.29
Trikut GOH	Machining Lubricant/Coolant	Pemaco, Inc.	21.1.20
Trikut HO	Machining Lubricant/Coolant	Pemaco, Inc.	21.1.31
Trikut HOL	Machining Lubricant/Coolant	Pemaco, Inc.	21.1.32
Trikut OBH	Machining Lubricant/Coolant	Pemaco, Inc.	22.1.32
Trikut OHD	Machining Lubricant/Coolant	Pemaco, Inc.	22.1.33
Trikut SHD	Machining Lubricant/Coolant	Pernaco, Inc.	22.1.35
	,		22.1.35
Trikut TPO	Machining Lubricant/Coolant	Pemaco, Inc.	33.1.7
Tuck 90W	Tape, Temporary	Tuck Tape Co.	25.1.4
Turco 520	Paint	Turco Products, Inc.	
Turco 379-40C	Protective Coating	Turco Products, Inc.	27.1.26
Turco 379-40D	Protective Coating	Turco Products, Inc.	27.1.27
Turco 4316	Scale Conditioner	Turco Products, Inc.	29.1.3
Turco 4338	Sca Conditioner	Turco Products, Inc.	29.1.4
Turco Pretreat	Prc tive Coating	Turco Products, Inc.	27.1.28
Type A Red	Mailing, Temporary	Magic Marker Corp.	23.2.6
Type A Red	Marking, Temporary	Speedry Chemical Prod. Co.	23.2.8
Vactra No. 4	Machining Lubricant/Cootant	Mobil Oil Co.	21.1.22
Vactra Oil Light	Machining Lubricant/Coolant	Mobil Oil Co.,	21.1.23
Vantrol 31-130A	Machining Lubricant/Coolant	Vantrol Corp.	21.1.44
Velocite No. 5	Machining Lubricant/Coolant	Mobil Oil Co.	21.1.24
Virgo	Scale Conditioner	Hooker Chem, Co.	29.2.3
W1 Blue, Rigidax	Stabilizer, Machining	M. Argüeso & Co., Inc.	31.1.3
WI Green, Rigidax	Stabilizer, Machining	M. Argueso & Co., Inc.	31.1.4
W1 NF, Rigidax	Stabilizer, Machining	M. Argüeso & Co., Inc.	31.1.5
Warren No. 1	Protective Coating	Warren Chemical Mfg. Co.	27 2.1
Warren No. 2	Protective Coating	Warren Chemical Mfg. Co.	27.2.2
Water	Solvent		30.1.22
WD	Penetrant Inspection Fluid	Fluoro-Chek Co.	26.1.3

TABLE 3.—ALPHABETICAL INDEX OF MATERIALS TESTED (continued)

Material Name or Designation	Material Type	Vendor	Number Code
White Oil No. 3, NF	Machining Lubricant/Coolant	Standard Oil Co. of Calif.	21.1.42
(Chevron)			
WK 308X	Protective Coating	Acheson Colloids Co.	27.1.3
WP-1	Penetrant Inspection Fluid	Turco Products, Inc.	26.3.39
WP-167	Penetrant Inspection Fluid	Turco Products, Inc.	26.3.40
WP-167R	Penetrant Inspection Fluid	Turco Products, Inc.	26.3.41
XP Kalcote	Protective Coating	California Coatings, Inc.	27.1.6
Y 9241	Tape, Temporary	3M Corp.	33.1.2
YF, Rigidax	Stabilizer, Machining	M. Argüeso & Co., Inc.	31.1.6
ZE-3	Penetrant Inspection Fluid	Magnaflux Corp.	26.2.2
ZE-4	Penetrant Inspection Fluid	Magnaflux Corp.	26.2.3
ZE-4A	Penetrant Inspection Fluid	Magnaflux Corp.	26.2.4
ZE-4B	Penetrant Inspection Fluid	Magnaflux Corp.	26.2.5
ZE-6 (ZR-1)	Penetrant Inspection Fluid	Magnaflux Corp.	26.2.6
ZL·2	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.2
ZL-2A	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.3
ZL-16	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.4
ZL-17	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.5
ZL-17A	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.6
ZL-18	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.7
ZL-22	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.8
ZL·22A	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.9
ZL-221	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.10
ZI30	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.11
ZL-30A	Penetrant Inspection Fluid	Magnaflux Corp.	26.3.12
ZP-4	Penetrant Inspection Fluid	Magnaflux Corp.	26.1.4
ZP-4A	Penetrant Inspection Fluid	Magnaflux Corp.	26.1.5
ZP-5	Penetrant Inspection Fluid	Magnaflux Corp.	26.1.6
ZP.9	Penetrant Inspection Fluid	Magnaflux Corp.	26.1.7
ZP-11	Penetrant Inspection Fluid	Magnaflux Corp.	26.1.8
ZP-20	Penetrant Inspection Fluid	Magnaflux Corp.	26.1.9
ZP·X432 (ZP·13)	Penetrant Inspection Fluid	Magnaflux Corp.	26.1.10
ZR-2	Penetrant Inspection Fluid	Magnaflux Corp.	26.2.7

5.3 TABULATION OF TEST DATA

Experimental test data is listed in table 4, pages 30 to 83, arranged in order of index numbers. Notations used in the table are as follows:

• Test. The various tests used are described in section 3 of this report.

• Test Parameters

Titanium alloys are identified as follows:

CP Commercially pure (unalloyed)

4A1-3Mo-1V or 4-3-1

5A1-2.5Sn or 5-2.5

6A1-4V or 6-4

8 Al-1Mo-1V or 8-l-1 where A1 = aluminum, etc.

Titanium alloy heat-treat conditions are designated as follows:

ndition I mill anneal (According to XBMS7-158 for
and 8-1-1)
ution Treated (ST)
ution Treated, Aged at 1000°F
ution Treated, Aged at 1250°F
plex Anneal

Specimens

Thickness was 0.050-inches unless otherwise specified. Dimensions according to directions for specific test. Number of specimens per test as follows (Minimum of two control specimens per test), unless otherwise specified in table 4.

<u>Test</u>	No. of Specimens (Per exposure condition per alloy)
Allison Bend	5 or more
Emittance	3
Flat Sheet Exposure	(Controlled by Allison Bend)
Gas Analysis	3
Resin Kettle	2
U-Bend (Ambient or Elev. Temp.)	2

Results (See section 4 for more detail)

U-Bend - Failure - Specimen broke into two parts

SCC - Stress-corrosion cracking observed*

No SCC or surface effect - Surface examined microscropically showed no attack, no metallographic examination done

^{*}For definition of numerical levels see table 1.

• Compatibility

- A = Compatible under test conditions employed, e.g. no excessive surface etch, SCC, or embrittlement of alloy tested.
- X = Incompatible, e.g. excessive etch, SCC, or embrittlement observed.

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS

			Г		_						_			٦			Π		_	
Compatibility Rating	-	∢	4	A (at operating	temperature,						· 2-	**	* *		₹			۷		
Results-Observations		Superficial etch	Slight surface etch		No visible surface	Bend energy, In Ibs	7	Gas content, ppm	H ₂ 0 ₂		No intergranular attack	Slight surface etch	Severe etch and pits		Diatomaceous earth, a long-chain hydrocarbon, and a carbonylingun	No SCC or surface effect			No SCC or surface effect	No SCC or surface effect
Test Parameters	Material Class Cleaner	75°F, 768 Hours	1000 F, 4 Hours	000	1/0 F, 1 Hr, Single	170°F, 1 Hr, 3 speci-	mens, 0.40"	170°F, 1 Hr.			170°F, 1 Hr.	1000°F, 4 Hr.,	Resin Kettle 1400°F, 45 Min.	Resin Kettle		850°F, 4 Hrs.	Material Class Food	c	850°F, 8 Hr.	850°F, 8 Hr.
Alloy	Materia	6-4 -	6-41		6-4	6-4 (6-4 1			64	6-41	6-4 1			6.4 1	Materi		4.3.11	8-1-11
Test		Single U-Bend	Single U-Bend		Flat Sheet	Allison Bend		Gas Analyein		_	Metallograph	Single U-Bend	Single U-Bend		Chemical Analysis (IR)	Single U.Bend			Resin Kettle	(35 ml milk)
Material		Pasa Jell 107C-7 (Semco Sales and Service Co.)	Bon-Ami	GMC 895	(Greater Mountain							Major Clean			Nevr-Dull (George Basch Co.)			Milk	-	
Material Number		18.1.1	18.1.2	18.1.3								18.1.4			18.1.5			19.1.1		

*Approved for use only if residual material is completely removed.

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		×	×	×	×	A			·×									•		
Results-Observations		ching and	ching and	grooving Etching	Circle of etching	t = 0.56	No surface effect				Gas Content, ppm	H ₂ ol 39 1	36 36 5	Control 30 1165	23 26	No microstructure	ditelations	Gas Content, ppm	Control 87 1020 Test 53 739	
Test Parameters	Material Class Heat Treating Aid	1475°F, 3 Hrs. as	1475°F, 3 Hrs. 1:1 H20	1400°F, 30 Min., as	received 1400°F, 30 Min., 1:1 H2O	1300 ⁰ F, 15 Min., 84 Ksi	stress 1000 ^o F, 4 Hrs.	Material Class Machining Lubricant/Coolant			Metal from inside	drilled holes	Metal from milled	Specimens Matal from specimens	following grinding	On each of above	specimen (Appea	Metal from inside		
Alloy	Material Cla	6.4	6-41	6-4	6-4 1	641	6.41	ial Class Ma		8.1.11		•	-	- , -			8.1.1 V			aniun.
Test		Single U-Bend	Single U-Bend	Single U-Bend	Single U-Bend	Heimerl-Braski	Single U-Bend	Mater		Machine Shop	1. Gas Analysis					2. Metallographic	Alidiysis	1. Gas Analysis		l It to corrosivity for tit
Material		Boron nitride				Fiberfrax			Barium Hydroxide (5% Solution)											Incompatible due to toxicity, not to corrosivity for titanium.
Material Number		20.1.1				20.1.2			21.1.1											Incompatit

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility	funte									×			ď																		
Test Parameters Results Observations			ď		1 87 1	Control 87 1020	Test 69 800	No microstructure	alterations	Severe surface etch						Gas Content, ppm	H ₂ O ₂		30 6	ς .	ccc 61 ISA1	No microstructure	alterations	Gas Content, ppm	H ₂ 0 ₂	_	Test 49 747	Control 87 1020	Test 51 686		
Test Parameters		Material Class Machining Lubricant/Coolant	Metal from milled	specimens		Metal from specimens	following grinding	On each of above	specimens	1000°F, 4 Hrs	(2 specimens)						Metal from inside	drilled holes	Motifican month	wetal from miled	specimens	On each of above	specimen types	Metal from inside	drilled holes			Metal from milled	specimens		
Allov		al Class Mac								6-41	_				8.1.11								8.1.1 V	-							
184]	- 1	Materi						2. Metallographic	Analysis	U-Berid					Machine Shop	Operations	1. Gas Analysis					2. Metallographic	Analysis	1. Gas Analysis							
Material			Barium Hydroxide	(5% Solution)						Cimcool T-10	(Cincinatti Milling	Machine Co.)	Freon-Butyl	Cellosoive	(DuPont Co.)																
Material	Number		21.1.1	(cont'd)						21.1.2			21.1.3																		

TABLE 4 -- COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		×					,	×			V		×		*	<		
Results-Observations			3.2% chlorine No SCC or surface effect	Specimens broke	No visible surface effect	Gas Content, ppm	H ₂ 0 ₂ Control 55 850 Test 60 815	SCC, Level 1			No effect		SCC, Level unrecorded				Contains attacked	solvents
Test Parameters	Material Class Machining Lubricant/Coolant		Bosler Method 850°F, 8 Hrs.	850°F, 8 Hrs.	550°F, 40 Min.	550°F, 40 Min.		1000°F, 4 Hr.			1000°F, 4 Hr.		1000°F, 4 Hr.				From company literatura	
Alloy	at Class Mac		4.3.11	6.4 1	 	8-1-11		6-4			6-41		6.4	-				
Test	Materi		Chloride Analysis Resin Kettle (43 ml coolant)		Flat Sheet	Exposure Gas Analysis		Single U-Bend			Single U-Bend		Single U-Bend				Chloride	
Material		Habcool 318 (H & B Petroleum Co.)						Crystal Cut	No. 322	(Hangsterfer's Laboratories, Inc.)	Missile Lube	(Hangsterfer's Laboratories, Inc.)	Missile Lube	(Hangsterfer's	Cut-Max 563	(E.F. Houghton	& Co.)	
Material Number		21.1.4						21.1.5		_	21.1.6		21.1.7		21.18			

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		*									A													_	
Results-Observations			ğ	H2 39	Test 31 726	alterations		Gas Content, ppm	H2 02	8 8		None defected	No mich occident	NO VISIBLE SULFACE ETTECT	المريس المستوال	Source Chiefly, man	Control 83	Test 59	ت	•	Bend Energy, In-Ibs		Control 98	Test 92	
Test Parameters	Material Class Machining Lubricant/Coolant		Metal from milled	specimens	A. (1)	specimens				Metal from milled		Dodos Modes	DONE WELLIOU	SSU F, 10 Hrs.	City of the second second	corrosion test	(6 specimens	0.040" thick)	550°F, 40 Min.		Specimens from flat	sheet exposure	(8 specimens,	0.040" thick)	
Alloy	I Class Mac	8.1.11					8-1-1 V						-	- 1	. 7 3	-			6-4!		6-41				
Test	Materi	Machine Shop Operations	1. Gas Anatysis		-	2. metallographic Analysis		1. Gas Analysis		2. Metallographic Analysis		Obligation Application	Sicyndia : Sicyndia :	Using Corresion	Test	Trian losing			Flat Sheet	Exposure	Ahisan Bend				
Material											Hocut 237 (E.F.	Houghton & Co.)													
Mate: ial Number		21.1.8 (cont'd)								 	21.19														

TABLE 4.-COMPATIBILITY DATA, MANUF. :TURING AID MATERIALS (continued)

Compatibili', Rating							-									-			-					A			
Results-Observations		Gas Content, ppm H2 02 Control 55 850	3	Gas Content, ppm	H ₂ O ₂ Control 39 1165	33	Test 50 850		Test 19 680		No microstructure	alterations		ğ	H ₂ O ₂	rol 87 1	45	Test 7.4 660		Test 56 620		No microstructure	alterations	Slight surface effect	_		
Test Parameters	chinir oricant/Coolant	On specimens after Allison bend		Metal from inside	drilled holes		Metal f.om ground	specimens	Metal from milled	specimens	On each of above	specimen types		Metai from inside	drilled holes			Metal from ground	specimens	Metal from milled	specimens	On each of above	specimen types	1000°F, 4 Hrs.			
Alloy	Material Class Machinir	941	8-1-1	_							8.1.11		8-1-1 V											6.4 1			
Test	Materi	Gas Analysis	Machine Shop Operations	1. Gas Analysis							2. Metallographic	Analysis		1. Gas Analysis								2. Metallographic	Analysis	Single U Bend	: ;	1	
Material																								Houghto Safe	620 (Houghten	(co.)	
Material Number		21,1.9 (cont'd)																						21.1.10			

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		∢			×		A										4		A		¥		
Results-Observations				No SCC or surface effect	Severe SCC			dirition O	Meninia la marca affasa	NO VISIDIE SULTACE ETTECT	Dond Control	Define Eriergy, invites	Control 98 Test 98	nte	H ₂ O ₂	Control 55 850	C or surface	No SCC or surface effect	No SCC or surface effect	No SCC or surface effect		600 ppm Chloride No visible surface effect	
Test Parameters	Material Class Machining Lubricant/Coolant			850°F, 6 Hrs. 850°F, 6 Hrs.	1000°F, 4 Hrs.			3.33		550 F, 40 Min.		220 F. 40 MID.,	8 specimens	On Allison Bend speci-	mens (2) after testing		850°F. 1.5 Hr.	850°F 15 Hr	850°F, 2 Hr.	850°F, 2 Hr.		Bosler Method 550°F, 40 Min.	
Alfoy	al Class Mac		·	641	6.4 1			_		4		0.4.0		4			64	2.1.1	6.41	8-1-11		641	
Test	Materi	Resin Kettle			U-Bend				Chloride	Flat Sheet	Exposure	Allison Beng		Gas Analysis			Resin Kettle		Resin Kettle			Chloride Flat Sheet	
Material		lodine (1%)	SAE 10 (95%)		Aquamatic Univer-	sal Coolant (Molecular Products, Inc.)	C:250	(Mobil Cil Co.)									DTF No 24	(Mobil Oil Co.)	DTE No. 26	(Mobil Oil Co.) BAC No. 12	MET 25 (Mobil Oil Co.)		
Material Number		21.1.11			21.1.12		21.1.13										21 1 14		21.1.15		21.1.16		

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility				٨		-		A					∢				⋖	∢								-
Results-Observations			Bend Energy, In-Ibs Control 83 Test 84			No SCC or surface effect	No SCC or surface affect			1.7% sulfur	No SCC or surface effect	No SCC or surface effect			No SCC or surface effect	No SCC or surface effect	No SCC or surface effect				Dod		Control 39 1165	rostructur	alterations	
Test Parameters		Material Class Machining Lubricant/Coolant	On flat sheet specimen (6)			850°F, 2 Hr.,	2 specimens			Vendor Literature	850°F, 3 Hr.	850°F, 3 Hr.			850°F, 2 Hr.	850 F, 2 Hr.	1000°F, 4 Hrs.				Metal from inside	drilled holes		On above metal		
Alloy		al Class Mad	64		-	- 49				-	6-4	8-1-11		1.	6-4	8.1.1.1	6-4			8-1-11				8-1-11		
Test Alloy Test Param	- 1	Materia	Allison Bend			Resin Kettle	(35 ml sample)			Chemical Content	Resin Kettle				Resin Kettle		U-Bend			Machine Shop	1. Gas Analysis			2. Metallographic	Analysis	
Material				MIST No. 36	(Mobil Oil Co.)	BAC NO. 25		Solvac 2032	(Mobil Oil Cc.)				Solvac NP	(Mobil Oil Co.)			Sultran 176-M (Mobil Oil Co.)	TJ.73	(Mobil Oil Co.)							
Material	Number		21.1.16 (cont'd)	21.1.17				21.1.18					21.1.19				21.1.20	21.1.21								

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating				4		ď	ď	ď	×	×
Results-Observations		Gas Analysis, ppm H2 O2 Control 87 1020 Test 43 564	cros		No SCC or surface effect No SCC or surface effect	ļ	No SCC or surface effect		Slight SCC and pits	SCC, Level No. 3
Test Parameters	Material Class Machining Lubricant/Coolant	Metal from inside <u>drilled</u> hole	Metal from milled specimens on metal from milled and drilled specimens		850°F, 1.5 Hr. 850°F, 1.5 Hr.	1000°F, 4 Hr.	850°F, 1.5 Hr. 850°F, 1.5 Hr.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.
Alloy	al Class Mac	8-1-1 V			<u>8</u>	64	641	64	6-4	6.4)
Test	Materi	1. Gas Analysis	2. Metallographic Analysis		Resin Kettle	U-Bend	Resin Kettle	U-Bend	U-Bend	Single U-Bend
Material				Vactra No. 4 (Mobil Oil Co.)		Vactra Oil Light (Mobil Oil Co.) BAC No. 2	Velocite No. 5 (Mobil Oil Co.) BAC No. 63	Cool Tool (Munroe Chem. Co.)	Coolube No. 10 (Pacific Chemical Co.)	Tricut DS (Pemaco, Inc.)
Material Number		21.1.21 (cont'd)		21.1.22		21.1.23	21.1.24	21.1.25	21.1.26	21.1.27

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		×	×	×	×	×	×	×	×	X	×														٨		ď			
Results-Observations		SCC, Level No. 3	SCC, Level No. 3	SCC, Level No. 3	SCC, Level No. 3	SCC, Level No. 3	SCC, Level No. 3	SCC, Level No. 3	SCC, Level No. 3	SCC, Level No. 3		. !	Contains 1, 1, 1-Tri-	chloroethane	No SCC	Specimens Broke	No visible surface effect		Bend Energy, In-Ibs	_	Test 90	Gas Content, ppm	H ₂ O ₂	Test 60 780	No. SCC or surface effect				Negative	
Test Parame	Material Class Machining Lubricant/Coolant	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.			Company Literature	c	850 F, 8 Hrs.	850°F, 8 Hrs.	550°F, 40 Min.		550°F, 40 Min.			On Allison Bend,	2 specimens		1000°F, 4 Hrs.				Beilstein	
Alloy	al Class Mac	- 40	6-41	6-41	641	641	6-4:	6-41	6-41	641					64 -	8-1-1	6-4		- 79			<u>4</u>			6-4 (_	
Test	Materi	Single U-Bend	Single U-Bend	Single U-Bend	Single U-Berd	Single U-Bend			Chloride		Resin Kettle		Flat Sheet	Exposure	Allison Bend			Gas Analysis			U-Bend				Chloride					
Material		Tricut DSL (Pemaco)	Tricut GO	Tricut GOH	Tricut HO	Tricut HOL	Tricut OBH	Tricut OHD	Tricut SHD	Tricut TPO	Rapid Tap	(Helton Corp.)													Eagle Oil No. 10	(Richfield Oil Co.)	Tapzol 410	(Rust-Lick Corp.)		
Material Number		21.1.28	21.1.29	21.1.30	21.1.31	21.1.32	21.1.33	21.1.34	21.1.35	21.1.36	21.1.37														21.1.38		21.1.39			

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating									A-X																			
Results-Observations		No visible surface effect	Bend Energy, In-Ibs	Control 98	Test 90	Gas Content, ppm	Н2	Control 55 850 Test 56 800		No SCC or surface effects	on initial testing. Suc-	sequent tests gave slight	to severe surface etch		-	Gas Content, ppm		_	Test 28 712	No microstructure	alterations	Gas Content, ppm	H2 02	_	Test 74 728	No microstructure	alterations	
Test Parameters	Material Class Machining Lubricant/Coolant	550°F, 40 Min.	Specimens (8) from	flat sheet exposure		On Allison Bend	specimens (4)			1000°F, 4 Hr.	-					Metal specimens after	grinding			On above specimens		Metal specimen after	grinding			On above specimens		
Alloy	al Class Mad	6-4								- 49						8-1-1	•			8-1-11	8-1-1 V					8.1.1 V		
Test	Materi	Flat Sheet	Allison Bend			Gas Analysis				U-Bend				Machine Shop	Operations	1. Gas Analysis				2. Metallographic	Analysis	1. Gas Analysis				2. Metallographic		
Material									Sodium Nitrite 5% solution of C.P.																			
Material Number		21.1.39	in allow					-	21.1.40																			

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		×			-						∢							-	-					
Results-Observations			Contains 1, 1, 1-Tri- chloroethane	No visible surface effect	Bend Energy, In-Ibs	Control 98	ಕ್ಷ	H ₂ O ₂	rol 55	Test 57 845			ē		-	Test 26 790	No microstructure	alterations	Gas Content, ppm	H ₂ 0 ₂	Test 73 668	No microstructure	alterations	
Test Parameters	Material Class Machining Lubricant/Coolant		Company Literature	550°F, 40 Min.	Flat sheet exposure	specimens	On Allison bend	specimens					Analysis on milled	specimens			On above specimens		Analysis on milled	specimens		On above specimens		
Alloy	at Class Ma			6-4 1	6-4 /		6-41					<u> </u>	8-1-11				8-1-1	8.1.1 V				> 		
Test	Materia		Chloride	Flat Sheet	Exposure Allison Bend		Gas Analysis				Machine Shop		1. Gas Analysis				2. Metallographic	Analysis	1. Gas Analysis			Z. Metallographic	Analysis	
Material		Tap Magic (Staco Corp.)									White Oil No. 3 NF	of California)											-	
Material Number		21.1.41									21.1.42													

TABLE 4.- COMPATIBILITY DATA, MANUFACTURING AID MATFRIALS (continued)

	Rating		4							4			×		×	×	×	×
IABLE 4 COMPATIBILITY DATA, MANOTACTORING AID MATERIALS COMMISSION	Results-Observations			800 ppm Chloride No visible surface effect		Bend Energy, In-Ibs	Control 98	onte	H ₂ 0 ₂ Control 55 850 Test 54 830	No surface effect			SCC, Level 1		SCC, Level 1	SCC, Level 1	SCC, Level 3	SCC, Level 2
מושושה האוועה האוועה	Test Parameters	Material Class Machining Lubricant/Coolant		Bosler method 550°F, 40 Min		On flat sheet exposure	specimens	On Ailison specimens		1000°F, 4 Hrs.		Material Class Marking, Electrochemical, Cleaner	1000°F, 4 Hrs.		1000°F, 4 Hrs.	1000°F, 4 Hrs	1000°F, 4 Hrs.	1000° F, 4 Hrs.
H H H H	Alloy	al Class Mac		6-4						6-4		Class Marki	6.4		6-4	6-4	6.4	6-4
JIMER HEILTH I D	Test	Materi		Chloride Flat Sheet	Exposure	Allison Bend		Gas Analysis		Single U-Bend		Material	Single U-Bend		Single U-Bend	Single U-Bend	Single U-Bend	Single U-Bend
HABLE 4 CO	Material		Grind Tex 8-410 (Texaco Co.)							Vantrol 31-130A	(Vantrol Corp.)		Neutracleaner	No. 1 (Electro Chem Etch Co.)	2.1 (Electromark Co.)	2.2 (Electromark	5 (Marking Methods, Inc.)	APC (Marking Methods, Inc.)
	Material Number		21.1.43							21.1.44			22.1.1		22.1.2	22.1.3	22.1.4	22.1.5

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TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		×	×		×	A	×	×	×	Ą	×	×	×	×	×
Results-Observations		SCC, Level 2	SCC, Level 1	te	SCC, Level 1	Slight etch, Level 0	SCC, Level 4	SCC, Level 4	SCC, Level 5	No surface effect	Etch, Level 0	SCC, Level 1	SCC, Level 1	SCC, Level 1	SCC, Level 1
Test Parameters	Material Class Marking, Electrochemical, Cleaner	1000°F, 4 Hrs.	1000° F, 4 Hrs.	Material Class Marking, Electrochemical, Electrolyte	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000 ³ F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.
Alloy	Class Markin	6-41	6-4 -	ass Marking	6-4	6.41	641	6-4 1	6-4 (6-4	6-4	6-4 1	6-4 1	6.4 1	6-4
Test	Material	Single U-Bend	Single U-Bend	Material C	Single U-Bend	Single U-Bend	Single U-Bend	Single U-Bend	Single U.Bend	Single U-Bend	Single U-Bend	Single U.Bend	Single U-Bend	Single U.Bend	Single U.Bend
 Material		MSC No. 3 (Monode, Inc.)	MSC No. 5 (Monode, Inc.)		359L (Electro Chem Etch Co.)	E S and T (Electro Chem Etch Co.)	1 (Electromark Corp.)	2 (Electromark Corp.)	4 (Electromark Corp.)	59-L (Electromark Corp.)	353 (Electromark Corp.)	SC-44 (Electro- mark Corp.)	SC-82 (Electro- mark Corp.)	T-10 (Marking Methods, Inc.)	B-10 (Manode, Inc.)
Material Number		22.1.6	22.1.7		22.2.1	22.2.2	22.2.3	22.2.4	22.2.5	22.2.6	22.2.7	22.2.8	22.2.9	22.2.10	22.2.11

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

	Compatibility Rating		*	×		4 4	×	×		٨	Ą	¥	۷	Ą	۷	
	Results-Observations	a.	SCC, Level 4	SCC, Level 1	ator	No surface effect No surface effect	Surface Roughened	Surface Roughened		No surface effect	No surface effect	No surface effect	No surface effect	No surface effect		No detectible chlorine
	Test Parameters	Material Class Marking, Electrochemical, Electrolyte	1000°F, 4 Hrs.	1000°F, 4 Hrs.	Material Class Marking, Temporary, Heat Treat indicator	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1350°F, 4 Hrs.	Material Class Marking, Temporary, Ink	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.		Beilstein
, , , , , , , , , , , , , , , , , , ,	Alloy	iss Marking,	6-4 1	6-4	s Marking,	6-41	6-41	6-4	rial Class M	6.4	6.4	6.4	6.4	6.4		
ייון און און וייון און	Test	Material Cla	Single U-Bend	Single U-Bend	Material Clas	Single U-Bend Single U-Bend	Single U.Bend	Single U.Bend	Mate	Double U-Bend	Single U.Bend	Single U-Bend	Single U.Bend	Single U.Bend		Chlorine
700114-100	Material		F.10 (Monode,	F.20 (Monode, Inc.)		"1150 F" (Tempilstik Co.)	"1350 F"	(Tempilstik Co.)		1107 NH (Dickson Ink Co.)	1307 (Dickson Ink Co.)	2101 (Dickson Ink Co.)	2307 (Dickson Ink Co.)	5507 (Dickson Ink Co.)	Type A Red Translucent (Magic Marker	
	Material Number		22.2.12	22.2.13		23.1.1	23.1.2			23.2.1	23.2.2	23.2.3	23.2.4	23.2.5	23.2.6	

TABLE 4 -- COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating						×		y		_	`,	K		A		197 (4	A	i i			.		
Results-Observations		Solvent system: 99%	No surface effect	No SCC or surface effect	No SCC or surface effect		6500 ppm chloride	Solvent system: 40% ethanol, 16% ethoxy-	ethanol, no chlorinated	solvents	Severe surface pits and SCC	No su: face effect		Either no SCC or surface	effect, or very slight	surface roughening (purple)	No SCC or surface effect	Slight surface roughening	or no SCC or surface	effect (blue, purple)	Secretary of Joseph	ואס סכר מו אחנושרה הנוהנו	-	
Test Parameters	Material Class Marking, Temporary, Ink		75°F, 24 Hrs.	1000°F 4 Hrs	1000°F, 4 Hrs.		Radiotracer Silver				1000°F, 4 Hrs.	1000°F, 4 Hrs.		1000°, 4 Hrs. Black,	brown, blue, green,	orange, purple, yellow tested	1000°F, 4 Hrs. Black,	1000°F, 4 Hrs. Black,	brown, blue, green,	orange, purple, yellow	1000 ² E 1 U.2 O.2222	Durnle vellow feeted		
Alloy	erial Class A		8-1-11	(undeaned) 6.4 I	8-1-11						- 4-9	641		641										
Test	Mat	IR Analysis	U-Bend				Chloride	IR Analysis			U-Bend	Double U-Bend		Single U-Bend										
Material						1001 Red (Pannier Ink Corp.)						Type A Red	(Speedry Chenical Products Co.)	Ink (Esterbrook	Co.)		M-3 (Marsh Co.)	Project-A-Mark	(Marsh Co.)		100000000000000000000000000000000000000	iiik (Siarpie co.)		T
Materia! Number		23.2.6	(collection)			23.2.7						23.2.8		23.2.9			23.2.10	23.2.11			22.2.13	23.2.12		

THELE 4 - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		ು ಎಸ್			. : . ; ; ; ;		- S 1.7			÷					.t.									
Compatib Rating		⋖			∢	×		⋖	A (?)	×	∢	×					⋖	A		∢	⋖			
Results Observations		No surface effect			No surface effect	Severe surface etch		No surface effect	Etch only	Massive surface effect	No surface effect	SCC, Level 5	-				No surface effect	Slight surface effect		No surface effect	No surface effect			
Test Parameters	Material Class Marking. Temporary, Pencil	1000°F, 4 Hrs.		Material Class Metal	<700°F, 168 Hrs.	>717°F, minimum time	not determined	500°F, 100 Hrs.	1000°F, 4 Hirs.	1650°F, 2 Hrs.	75°F, 69 Hrs.	320°F, 20 Min. titanium	heated to 1000°F,	45 Min. prior to stressing		Material Closs Paint, Leak Detector	1000°F, 4 Hrs.	1000°F, 4 Hrs.		1000°F, 4 Hrs.	1000°F, 4 Hrs.			
Alloy	ial Class Ma	6-4 1		Mater	1 0-9	1 5-9		- 4	6-4 !	641	6-4 !	6-41				aterial Class	6-41	6.41		6-4 1	9-41			
est	Mater	Single U.Bend			Double U-Bend	Double U-Bend		Double U-Bend	Double U-Bend	Double U-Bend	Single U-Bend	Single U-Bend				Ž -	Single IJ. Bend	Single U-Bend		Single U-Bend	Single U-Bend			
Material		55 i Blue	(Blaisdell Co.)		Kirksite			Lead			Mercury						Orange Detector	Ke'ite No. 9	(Kelite Chem Co.)	Snoop Nc. 3 (Nupro Co.)	Turco 520	(Turco Products,	luc.}	
Material Number		23.3.1			24.1.1			24.1.2			24.1.3						25.1.1	25.1.2		25.1.3	25.1.4			

TABLE 4 .- COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility	Rating		¥	3			<.	- 1	Artical Park Source	A	:	. 21 2 2 3 7 2 5					·	¥,	-		×	×	,	-	
	Results-Observations			No SCC or surface effect	No SCC or surface effect	Slight surface etch only		No SCC or surface effect	Slight surface etch only No SCC or surface effect				No SCC or surface effect	No SCC or surface affect	No SUC Of SUPACE ETTECT	er .	No surface effect		Slight etch at material-	titanium interface	Pitting	Severe etching and	pitting		- (
	Test Parameters	Material Class Paint, Protective		450°F, 168 Hrs.	500°F, 168 Hrs.	1000°F, 4 Hrs.		450°F, 168 Hrs.	600°F, 168 Hrs. 1000°F, 4 Hrs.			,	450°F, 168 Hrs.	600°F, 168 Hrs.	1000 F, 4 HIS.	Material Class Penetrant Inspection Fluid, Developer I	1000°F, 4 Hrs,	Dry Powder	1000° F, 4 Hrs.	Water slurry	1000°F, 4 Hrs.	1000 F, 4 Hrs.			
	Alloy	Material Ci		- 49	641	6-4		į γ.9	6-4				6-4	2 4 5	Ĭ	lass Penetra	6.41		6-41		641	5.4 1			
	Test			Single U-Bend	Single U-Rend	Single U-Bend		Single U-Bend	Single U-Bend Single U-Bend				Single U.Bend	Single U-Bend	Single O-bend	Material C	Single U.Bend		Single U.Band		Single U.Bend	Single U-Band			
	Material		Epoxy Primer,				Epoxy Enamel (BMS 10-11)			Polyurethane	Enamel	(BMS 10-11)					9-D-3 (Androx,	Australia PTY,			DD-2 (Fluro- Chek Co.)	'VD (Fluro-	Chak Co.)		
Material	Number		25.2.1				25.2.2			25.2.3							26.1.1				26.1.2	26.1.3			

=======================================	Compatibility Rating		×		4	×	- 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1			×		=	. <u>-</u>	4		×	A		×	ī. s.	×	1 1	N. Fr	٨	-
COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)	Results-Observations	ber -		167 opm Cl Slight surface etch	No surface effect		134 ppm CI	Slight surface etch	Singnt Pich, Level O			298 ppm CI	Slight surface etch	No surface effect		SCC, Levei No. 1	No surface effects		Severe surface etch			5 ppm Cl	Mild surface spots and etch	Very slight etch, Level C	
NUFACTURING AID A	Test Parameters	Material Class Penetrant Inspection Fluid, Developer		Bomb method	1000°F, 4 Hrs.		Bomb method	1000 F, 4 Mrs.	IOOU T, 4 AIS.			Bomb method	1000°F, 4 Hrs.	1000°F, 4 Hrs.		1000°F, 4 Hrs.	1000°F, 4 Hrs.		1000°F, 4 Hrs.			Bomb method	1000°F, 4 Hrs.	1090°F, 4 Hrs.	
4 TA, MA	Alloy	ass Penetra		6.4	6.41			641	0-4				6-4	1 7-9		641	6.41		8-1-11				×1:1.20	6.4 1	
OMPATIBILITY D	Test	Material C		Chloride Sixele (1.Read	Single U-Bend		Chloride	U-Bend	Single U-Bend			Chioride	U-Bend	Single U-Bend		Single U-Band	Single U-Bend		U-Bend			Chloride	U.Bend	Single U-Bend	-
1ABLE 4CC	Material		ZP.4		ZP-4A	ZP.5 (Magnaflux Corp.)		007	(Magnaflux Corp.)	ZP.11	(Magnaflux Corp.)			ZP.20	(Magnaflux Corp.)	ZP.X-432 (ZP-13) (Magnaflux Corp.)	6.70	(Met-L-Chek Co.)	D 492-B (Shannon	Luminous Mater- ials Co.)	D-492-C (Shannon)			D 493 (Shannon)	
	Material Number		26.1.4		26.1.5	26.1.6		26.17	7.1.67	26.1.8				26.1.9		26.1.10	76.1.11		26.1.12		26.1.13			26.1.14	

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating			<u> </u>						-															
Compatib Rating		⋖		4			×	٧	٨	₫		×	×		٧		∢			×	×			
Results-Ouservations	er	Ç	283 ppm CI No SCC or surface effect		123 ppm Cl	No SCC or surface effect	Roughened surface	No surface effect	No surface effect	No surface effect		Severe etch	Rough etching and pit-	ting. Retest gave SCC, Level 2	No surface effect				103 ppm CI No SCC or surface effect	Slight surface etch		200 ppm CI	Slight surface etch	
Test Parameters	Material Class Penetrant Inspection Fluid Developer		Bomb method 1000°F, 4 Hrs.		Bomb method	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs	1000°F, 4 Hrs.	1000°F, 4 Hrs.		1000°F, 4 Hrs.	1000°E, 4 Hirs.		1000°F, 4 Hrs.				Bomb method 1000°F, 4 Hrs.	1000 [°] F, 4 Hrs.			1000°F, 4 Hrs.	
Alloy	ass Penetrar		8.1.11			8-1-11	6-4	6-4	6-4 1	6-4 1		8.1.1	6-41		641				8.1.1	8.1.1			8.1.1	
Test	Material CI	:	Chloride U.Bend		Chloride	U-Bend	Single U-Bend	Single U-Bend	Single U-Bend	Single U-Bend		Single U-Bend	Single U-Bend		Single U Bend				Chloride U-Bend	U-Bend		Chloride	U.Bend	
Material		D-493A (Shannon)		D-495A (Shannon)			D-499B (Shannon)	D-499C (Shannon)	D-90 (Sherwin, Inc.)	D-100 (Sherwin,	Inc.)	D-110 (Sherwin, Inc.)	D-113A (Sherwin,	Inc.)	L-028 (Sherwin,	(nc.)	DD-535 (Sperry Div., Automated	Ind., Inc.)		DW-530 (Sperry Div.)	FS-33 (Testing	Systems, Inc.)		
Material Number		26.1.15		26.1.16			26.1.17	26.1.18	26.1.19	26.1.20		26.1.21	26.1.22		26.1.23		26.1.24			26.1.25	26.1.26			

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

	Compatibility Rating		×	×	٨				٧		۷		⋖	A	A		-		
COMPATIBILITY DATA, MANOCACTONING AID IMATERIALS (COMMINGED)	Results-Observations	er	cc	Slight SCC		26 ppm Cl	No SCC or surface effect	No SCC or surface effect	140 ppm Cl No SCC or surface effect	emover	No surface effect		No surface effect	No surface effect		176 ppm Cl	No SCC or surface effect	No SCC or surface effect	No SCC or surface effect
NOTACIONING AID IN	Test Parameters	Material Class Penetrant Inspection Fluid, Developer	1000°F, 4 Hrs.	1000°F, 4 Hrs.		Bomb method	850°F, 8 Hrs.	850 F, 8 Hrs.	Bomb method 1000°F, 4 Hrs.	Material Class Penetrant Inspection Fluid, Emulsifier/Remover	1000°F, 4 Hrs. 5%	water solution of material	1000°F, 4 Hrs. 20%	1000°F, 4 Hrs.		Bomb method	1000°F, 4 Hrs.	850°F, 8 Hrs.	850°F, 8 Hrs.
4/A, IMA	Alloy	ass Penetrar	8-1-11	8-1-11		707	- 1 6	8.1.11	8-1-11	enetrant In	6-4 1		6-4 1	6-41			8-1-1-1	4.3.11	6.41
WEATIBILLE D	Test	Material Cl.	U.Bend	U-Bend		Cinloride	Resin Kettle	Resin Kettle	Chloride U-Bend	Material Class P	Single U.Bend		Single U-Band	Single U-Bend		Chtoride	U-Bend	Resin Kettle	Resin Kettle
MARKE 4CO	Material		Fluoro Finder Dry (Testing Systems, Inc.)	Fluoro Finder Wet (Testing Systems, Inc.)	Dy-Chek (Turco Products, Inc.)		•		Nad Developer (Turco Products)		9 PR-4 (Ardrox	Australia P.I.Y. Ltd.)		ZE-3 (Magnaflux)	ZE-4 (Magnaflux)				
	Material Number		26.1.27	26.1.28	26.1.29				26.1.30		26.2.1			26.2.2	26.2.3				

TABLE 4.—COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		4	A	A (?)					,	×								A		A		×			×			A	A	
Results-Observations	mover	No surface effect	No surface effect		119 ppm Cl	Pits and surface etch		No SCC or surface effect				51 ppm Cl	Severe SCC		SCC and pits			No surface effect		No surface effect		SCC, Level 1				384 ppm Cl	Slight surface etch	No surface etch	Very siight etch, Level 0	
Test Parameters	Material Class Penetrant Inspection Fluid, Emulsifier/Remover	1000°F, 4 Hrs.	1000°F, 4 Hrs.		Bomb method, Batch "X"	1000°F, 4 Hrs. Batch	"X" 5% solution in c.w.	1000°F, 4 Hrs. Batch	. DA SOLUTION IN C.W.			Bomb method	1000°F, 4 Hrs. 5%	solution in c.w.	1000°F, 4 Hrs. 5%	solution in distilled	water	1000°F, 4 Hrs.		1000 ³ F, 4 Hrs.		1000°F, 4 Hrs.				Bomb method	1000'F, 4 Hrs.	1000°F, 4 Hrs.	1000 F, 4 Hrs.	
Alloy	netrant Ins	6-4	1 49			6-41		9-41					64		8-1-11			6-4		1 4.9		6-4 (8.1.1.1	6.41	6.4 1	
Test	Material Class Pe	Single U-Bend	Single U-Bend		Chloride	U-Bend						Chloride	U-Send					Single U-Bend		Single U-Bend		Single U-Bend			:	Chloride	U-Bend	Single U.Bend	Single U-Bend	
Material		ZE-4A (Magnaflux)	ZE-48 (Magnaflux)	ZE-6(ZR-1) (Magnaflux)						ZR-2 (Same as ZE-6 plus antirust)	(Magnaflux)	•						E-56	(Met-L-Chek Co.)	E-59	(Met-L-Chek Co.)	E-142 (Shannon	Luminous Mater-	tals Co.)	E-153 (Shannon)			E-157A (Shannon)	ER-82 (Sherwin,	Inc.)
Material Number		25.2.4	26.2.5	26.2.6					. 0	26.2.7								26.2.8		26.2.9		26.2.10			26.2.11			26.2.12	26.2.13	

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		⋖	٧				ď		4		4	· •					٨			A	٧			
Results-Observations	emover	No SCC or surface etch		422 ppm Cl	No SCC or surface etch	No SCC or surface etch	No surface effect	וטנ	No surface effect			59 ppm Cl	No SCC or surface effect		75 ppm Cl	No SCC or surface effect	No surface effect		67 ppm Ci	No SCC or surface effect				
Test Parameters	Material Class Penetrant Inspection Fluid, Emulsifier/Remover	1000°F, 4 Hrs.		Bomb method	850°F, 8 Hrs.	850°F, 8 Hrs.	1000°F, 4 Hrs.	Material Class Penetrant Inspection Fluid, Penetrant	1000°F, 4 Hrs.			Bomb method	1000°F, 4 Hrs.	850°F, 8 Hrs.	850°F, 8 Hrs.	850°F, 8 Hrs.		Bomb method	1000°F, 4 Hrs.	1000°F, 4 Hrs.		Bomb method	1000°F, 4 Hrs.	
Alloy	netrant Ins	8-1-11			6.4.1	8-1-11	<u>4</u>	lass Penetra	641				6.4 1	4.3.1.1	6-4 1	8-1-11			641	6-4 1			6-4 1	
Test	Material Class Pe	U-Bend		Chloride	Resin Kettle Besin Kettle	Resin Kettle	Single U-Berd	Material C	Single U-Bend			Chloride	U-Bend	Resin Kettle	Resin Kettle	Resin Kettle		Chloride	U-Bend	Single U-Bend		Chloride	U.Bend	
Material		PE-520 (Sperry Div., Automated Ind., Inc.)	Dy-Chet. (Turco Products, Inc.)				E (Turco Products)		985-P2 (Ardrox	Australia PTY, Ltd.)	ZL-2 (Magnafiux)						ZL-2A (Magnaflux)			ZL-16 (Magnaflux)	ZL-17 (Magnaffux)			
Material		26.2.14	26.2.15				26.2.16		26.3.1		26.3.2			_			26.3.3			26.34	26.3.5			

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		⋖	۷		∢					٧	A			4			4				4	A	
Results-Observations	nt	No surface effect	730 ppm Cl	No SCC or surface effect		77 ppm Cl	No SCC or surface effect	No SCC or surface effect	No SCC or surface effect	No surface effect			73 ppm Cl No SCC or surface effect		105 ppm Cl	No SCC or surface effect			96 ppm Cl	No SCC or surface effect	Very slight surface effect	No surface effect	
Test Parameters	Material Class Penetrant Inspection Fluid, Penetrant	1000°F, 4 Hrs.	Bomb method	700 F, 3 Hrs. plus 1000°F, 4 Hrs.		Bomb method	850°F, 8 Hrs.	850°F, 8 Hrs.	850°F, 8 Hrs.	1000°F, 4 Hrs.			Bomb method 1000°F, 4 Hrs.		Bomb method	700°F, 3 Hrs. plus 1000°F, 4 Hrs.			Bomb method	700°F, 3 Hrs. plus 1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	:
Alloy	lass Penetra	6-4		4			43.11	4	8-1-11	641			4			<u>1</u>				<u>4</u>	6-4 1	641	
Test	Material C	Single U-Bend	Chloride	U-Bend		Chloride	C-bend Resin Kettle	Resin Kettle	Resin Kettle	Single U-Bend			Chloride U-Bend		Chloride	U-Bend			Chloride	U-Bend	Single U-Bend	Single U-Bend	
Material		ZL-17A (Magnaflux)	ZL-18 (Magnaflux)		ZL-22 (Magnaflux)					ZL-22A (Magnaflux)	ZL.221	(Magnaflux)		ZL-30 (Magnaflux)			ZL-30A	(Magnaflux)			FP.90 (Met-L-Chek)	FP.91	(Met-L-Chek)
Material Number		26.3.6	26.3.7		26.3.8					56.3.9	26.3.10			26.3.11			26.3.12				26.3.13	26.3.14	

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating												-																	
_	_	⋖	4	4		⋖	٨	-	∢		∢	ŀ	∢ —	_	<u>م</u>				۷	٧				۷	∢				
Results-Observations	ant	Slight surface effect	Slight edge etch	Slight edge etch		No SCC or surface effect	No SCC or surface effect		No SCC or surface effect		No SCC or surface effect	33 33 34	Slight etch effect			80 ppm Cl	No SCC or surface effect	No SCC or surface effect	No surface effect		148 ppm Ct	No SCC or surface effect		No surface effect		148 ppm CI	No SCC or surface effect		
Test Parameters	Material Class Penetrant Inspection Fluid, Penetrant	1000°F, 4 :Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.		1000°F, 4 Hrs.	1000°F, 4 Hrs.		1000°F, 4 Hrs.		1000°F, 4 Hrs.	00000	1000 F, 4 Hrs.			Bomb method	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.		Bomb method	700°F, 3 Hrs. pius	1000 F, 4 Hrs.	1000°F, 4 Hrs.		Bomb method	700°F, 3 Hrs. plus	1, 4 118.	
Alloy	lass Penetra	6.4 1	64	6-41		6-4	6-4 1		641		6-41		<u>\$</u>				64	8-1-11	6-4 1			6-4 1		6-4 1			6.4		
Test	Material C	Single U-Band	Single U-Bend	Single U-Bend		U-Bend	U-Bend		U-Bend		U-Bend		N-Bend			Chloride	U-Bend	U-Bend	Double U-Bend		Chloride	U-Bend		Double U-Bend		Chloride	U-Bend		
Material		FP-92 (Met-L-Chek)	FP-93	FP.95	(Met-L-Chek)	P3F (NAA. Bockwell)	P3F 3 (NAA	Rockwell)	P5F-1 (NAA-	Rockwett)	PSF-2.5 (NAA-	nockwell)	PSF-3 (NAA-	Rcckwell)	P.133 (Shannon)				P-133A (Shannon)	P.134 (Shannon)				P.134A	P.135 (Shannon)				
Material Number		26.3.15	26.3.16	26.3.17		26.3.18	26.3.19		26.3.20		26.3.21	2000	26.3.22		26.3.23				26.3.24	26.3.25				26.3.26	26.3.27				

TABLE 4 .-- COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		⋖	1±-1¥ .	A	۷	∢		А	٨	۵	4
Results-Observations	nt 	65 ppm Cl No SCC or surface effect	1% CI No SCC or surface effect	1% CI No SCC or surface effect	No SCC or surface effect	÷	44 ppm Cl No SCC or surface effect	No SCC or surface effect	No SCC or surface effect	55 ppm Cl	55 ppm Cl No SCC or surface effect
Test Parameters	Material Class Penetrant Inspection Fluid, Penetrant	Bomb method 700°F, 3 Hrs. plus 1000°F, 4 Hrs.	Bomb method 700°F, 3 Hrs. plus 1000°F, 4 Hrs.	Bomb method 700°F, 3 Hrs. plus 1000°F, 4 Hrs.	1000°F, 4 Hrs.	-	Bomb method 1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	Bomb method	Bomb method 700°F, 3 Hrs. plus 1000°F, 4 Hrs.
Alloy	lass Penetra	6-41	54	<u>-</u>	64 -		641	641	641		- 49
Test	Material CI	Chloride U-Bend	Chloride U-Bend	Chloride U-Bend	Single U-Bend		Chloride U-Bend	U-Bend	U-Bend	Chloride	Chloride U-Bend
Material		P 148A (Shannon)	P 149 (Shannon)	P 150 (Shannon)	FP-22 (Sherwin, Inc.)	HM-3 (Sherwin, Inc.)		HM-225 (Sherwin, Inc.)	HM-405 (Sherwin, Inc.)	FPE 505 (Sperry Div., Automated ind., Inc.)	FL.50
Material Number		26.3.28	26.3.29	26.3.30	26.3.31	26.3.32		26.3.33	26.3.34	26.3.35	26.3.36

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

tions Compatibility Rating		∢		e effect	e effect	٧		Ø			e offect	ď	=	e effect	۲		∢	٨	∢	ď	_
 Results-Observations	trant I		236 ppm CI	No SCC or surface effect	No SCC or surface effect	No surface effect				53 ppm Cl	No SCC or surface effect		78 ppm CI	No SCC or surface effect	No surface effect	_	Etch, Level 0	No surface effect	No surface effect	No surface effect	
Test Parameters	Material Class Penetrant Inspection Fluid, Penetrant		Bomb method	850°F, 8 Hrs.	850 F, 8 Hrs.	1000°F, 4 Hrs.				Bomb method	700 F, 3 Hrs. plus 1000 F, 4 Hrs.		Bomb method	700°F, 3 Hrs. plus 1000°F, 4 Hrs.	1000°F, 4 Hrs.	Material Class Protective Coating, Heat Treat	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	
 Alloy	lass Penetra			4-3-11	8.1.1	6-41					6-4			<u>4</u>	6-4	l Class Prot	6-4 1	1 7-9	6.41	6-41	
Test	Material C		Chloride	Resin Kettle	Resin Kettle	Single U-Bend				Chloride	N-Bend		Chloride	U-Bend	U-Bend	Materia	Single U-Bend	Single U-Bend	Single U-Bend	Single U-Bend	7 - 0 - 1 - 1 - 0
 Material		Dy-Chek (Turco Products)		•		P-40 (Turco	Products)	WP-1 (Turco	Products)			WP.167 (Turco Products)			WP-167R (Turco Products)		Delta 31 (Acheson Colloids Co.)	FH-420 (Acheson)	WK-308X (Acheson)	Boric Acid	
Material Number		26.3.37				26.3.38		26.3.39				26.3.40			26.3.41		27.1.1	27.1.2	27.1.3	27.1.4	

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

, -	Compatibility Rating		۷		٨						-					-			∢						
	Results-Observations		No surface effect			No visible surface effect	after descaling	No visible surface effect	after descaling	No visible surface effect	arter descaling	Bend Energy, in:lbs	Control 100	Test 83	Control 109	Test 98	Control 28	lest 30	Negative (Oralisative	only)		No visible surface effect	after descaling		
	Test Parameters	Material Class Protective Coating, Heat Treat	1450°F, 1 Hr.			1000°F, 4 Hrs. single	specimen	1350°F, 30 Min. single	specimen	1725°F, 10 Min. single	specimen	1000°F, 4 His.			1350°F, 30 Min.		1725°F, 10 Min.		Rocler method		1000°F, 4 Hrs.	1350°F, 30 Min.	1/25 F, 10 Min.		
	Alloy	I Class Prot	- 4-9			641		64		6-41	•	641			64	•	4				6-41	6-41	641	+	
	Test	Materia	Single U-Bend			Flat Sheet	Exposure	Flat Sheet	Exposure	Flat Sheet	Exposure	Allison Bend			Allison Bend		Allison Bend		Phloride		Flat Sheet	Exposure			
	Material		Boric Acid plus General Electric SR350 silicone	ethyl ketone	Kalcote XP (California Coatings Co.)	•													Kalube FL-40 (Kalgard FL-40) (California Coat- ings Co.)						
	Material Number		27.1.5		27.1.6														21.1.7	_					

TABLE 4.--COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

	Compatibility Rating						⋖				· · · · · · · · · · · · · · · · · · ·			- <i>높</i> *	- ;	
COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)	Results-Observations		Bend Energy, In-lbs Control 100	<u>0</u>	Control 28	70		Negative (Qual. only)	No visible surface effects	after descaling	inerg	5	- 3	Test 50 Control 7 Test 9		
NUFACTURING AID M	Test Parameters	Material Class Protective Coating, Heat Treat	1000°F, 4 Hrs.	1350°F, 30 Min.	1725°F, 10 Min.	1900°F, 10 Min.		Bosler method	1000 ⁰ F, 4 Hrs. 1350 ⁰ F, 30 Min	1725 ⁰ F, 10 Min. (1900 ⁰ F, 10 Min.)	1000°F, 4 Hrs.	1350°F, 30 Min.	1725°F, 10 Min.	1500°F, 10 Min.		
ATA, MA	Alloy	I Class Prot							641	6-41	6-41	6-4	6.41	6.4 (
OMPATIBILITY D	Test	Materia	Allison Bend					Chloride	Flat Sheet Exposure		Allison Bend	Allison Bend	Allison Bend	Allison Bend		
TABLE 4C(Material						Kalube FL (Kalgard FL) (California Coat- ing Co.)									
	Material Number		27.1.7 (cont'd)				27.1.8									

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating				2				2			· · ·		e e e e e e e e e e e e e e e e e e e	-,		27.5		V		Toronto.			-		'			<u> </u>	3.5	
tions	-		tance			0.15	0.15	0.15	0.15	0.15	0.15	0.19	0.18	0.175	0.22	0.20	0.18			;	effect			os o						
Results-Observations		Metal	Dayomay	Mils per	side	0.0	1.2	1.9	0.0	-	1.9	0.0	9.0	1.9	0.0	1.4	2.3			,	No visible surface effect	after descaling		Bend Energy, In-Ibs	rol 100	<u>5</u>	70	_	5	56
Resu			°F			1000			1350			1725			1900					;	No vis	after c		Bend	Control	Test	Control	Test	Control	Test
Test Parameters	Material Class Protective Coating, Heat Treat	Flat sheet exposure	Surface removed by	chemical milling per	BAC 5842 as indicated														C	1000°F, 4 Hrs.	1350°F, 30 Min.	1900°F, 10 Min.		1000°F, 4 Hrs.			1350°F, 30 Min.		1725°F, 10 Min.	
Alloy	Class Pro	4												_						6-41	0.4	6-41		6-41			6-41		6-41	
Test	Materia	Emittance																	i	Flat Sheet	Exposure			Allison Bend						
Material																		T-50 (Everlube	Corp. of America)											
Material Number		27.1.8	(cont.d)	-						- 					-			27.1.9												

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

	Compatibility Rating				- *		2											-		4			1.						-			-		
-	tions	n-lbs.				Emit	tance			0.16	0.15	0.15	0.15	0.15	0.15	0.19	0.155	0.20	0.185				effect		effect		ps sq						-	
	Results-Observations	Bend Energy, In-Ibs.		2	Metal	Removed		Mils per	side	0.0	<u>-</u> :	2.0	0.0	6.0	2.0	0.0	9.	0.0	1.8			۸e	No visible surface effect	after descaling	No visible surface effect	after descaling	Bend Energy, In-lbs	-			23			
	Resu	Benc	Control	Test		a	ц°			1000			1350			1725		1900				Negative	No visi	after d	No visi	after d	Bend F	Cont	Test	Control	Test			
	Test Parameters	Material Class Protective Coating, Heat Treat	1900°F, 10 Min.		Flat sheet exposure	specimens as above. Sur-	face removed by chemi-	cal milling per BAC	5842 as indicated													Bosler method	1100°F, 30 Min.		1100°F, 30 Min.	÷	1100°F, 30 Min.		c	1100 F, 30 Min.				
	Alloy	Class Prote	6.41								-	-											6.41		8-1-11		641			8.1:1-				
	Test	Material	Allison Bend		Emittance																	Chloride	Flat Sheet	Exposure			Allison Bend							
	Material															_				C-300	(Fel·Pro., Inc.)													
	Material Number		27.1.9	(cont'd)																27.1.10														

TABLE 4.--COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		-1					×		`.·		×		÷	-	×	3	77.9		×			Á	A (500°F max.)	-	2		×	- 1		· ·	er *
Results-Observations		Gas Content, ppm	H ₂	Test 57 1095	rol 45	Test 53 643	Ceramic coating left	after temperature expo-	sure, not removable by	nitric-fluoride etch	Ceramic coating left	after temperature expo-	sure, not removable by	nitric-fluoride etch	Cerainic coating left	after temperature expo-	sure, not removable by	nitric fluoride etch	Ceramic coating left	after temperature expo-	sure, not removable by	nitric-fluoride etch	Dull surface finish (char-	acteristic of effect pro-	duced by not spraying	material onto surface)		Speckled surface		Badly speckled surface	
Test Parameters	Material Class Protective Coating, Heat Treat	Flat sheet exposure	specimens		Flat sheet exposure	specimens	1000°F, 4 His.	1250°F, 4 Hrs.	1725°F, 30 Min.	1900°F, 30 Min.	1000°F, 4 Hrs.	1250°F, 4 Hrs	1725°F, 30 Min.	1900°F, 30 Min.	1000°F, 4 Hrs.	1250°F, 4 Hrs.	1725°F, 30 Min.	1900°F, 30 Min.	1000°F, 4 Hrs.	1250°F, 4 Hrs.	1725°F, 30 Min.	1900°F, 30 Min.	1000°F, 4 Hrs.					1000°F, 4 Hrs.		1250°F, 4 Hrs.	
Alloy	Class Prot	6-41			8:1:1		6-41				6-41				6-41				₽-9				8.1.11					6.4		6.4	
Test	Materia	Gas Analysis					Flat Sheet	Exposure			Flat Sheet	Exposure			Flat Sheet	Exposure			Flat Sheet	Exposure			U-3end					Flot Sheet	Exposure	Flat Sheet	Exposure
Material							A-498	(Glidden-Metiseci)			RA-536	(Glidden-Metiseel)			RA-537	(Glidden-Metiseet)			RA-536	(Glidden-Metlseel)			HD 200 (Kerns	Pacific Corp.)			HD205 (Kerns			«·	
Material Number		27.1.10	(cont'd)		-		27.1.11				27.1.12				27.1.13				27.1.14				27.1.15				27.1.16				

TABLE 4 - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		- 22 - 11	era gyar	Ą	. =	**	A		,,,-			,3 [.]				, T				<u>-</u>	<u>-</u>	-					A (600°F max.)
Results-Observations		Frost pattern on surface	Frost patteron on surface	Dull surface finish (char-	acteristic of effect pro-	duced by not spraying material onto surface)		Very slight duli surface	finish	Very slight duli surface	finish	Very slight dull surface	finish	Very slight dull surface	finish	No surface effect		Severe etch, with obvious	brush marks	Severe etch, with obvious	brush marks	1 Mil/side to remove dull	finish from 1000°F	specimens, 2.5 Mil/side	to remove etch from	1725 F specimens	Dull surface finish
Test Parameters	Material Class Protective Coating, Heat Treat	1725°F, 30 Min.	1900°F, 30 Min.	1000°F, 4 Hrs.				1000°F, 4 Hrs.		1000 F, 4 Hrs.	d	1000'F, 4 Hrs, Brush	application	1000°F, 4 Hrs, Brush	application	1000°F, 4 Hrs, Dip coat	application	1725°F, 10 Min.		1725°F, 10 Min.		Chemical milling accord-	ing to BAC 5842				1000°F, 4 Hrs.
Alfoy	Class Prote	6-4	6.4)	8-1-1-1	·			6-4		8.1.1		6-4		8.1.11		6-4		5-41		8.1.11		6-4	and	8-1-1-1			8-1-1
Test	Materia	Flat Sheet	Exposure Flat Sheet Exposure	U-Bend				U-Bend				Flat Sheet	Exposure					Flat Sheet	Exposure			Metal Removal					U:Bend
Material				HD6330	(Kerns Pacific)		LF22 (Lubri-Film Inc.)																				LF:32 (Lubri-Film, Inc.)
Material Number		27.1.16	(cont'd)	27.1.17			27.1.18																				27.1.19

TABLE 4 .-- COMPABILITY DATA, MANUFACTURING AID MATERIALS (continued)

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	Compacibility Rating		×	×	×	A (600°F max.)	A (RGU ^o F max.)	A (630°F max.)	A	4	∢							
TERIALS (continued)	Results-Observations		Did not clean up	Did not clean up	Did not clean up	Very slight dull surface effect	Dull surface finish	Very slight dull surface finish	No surface effect	No surface effect		No visible surface effect	after descaling No visible surface effect	after descaling No visible surface effect	after descaling No visible surface effect	after descaling Bend Energy, In-Ibs	Control 100 Test 104	Control 109 Test 107
COMPABILITY DATA, MANUFACTURING AID MATERIALS (continued)	Test Parameters	Material Class Protective Coating, Heat Treat	1450°F, 1.5 Hrs.	1450°F, 1.5 Hrs.	1450°F, 1.5 Hrs.	1000°F, 4 Hrs.	1000°F, 4 Hrs.	1000 ^c F, 4 Hrs.	1450°E, 1.5 Hrs.	1450°F, 1.5 Hrs.		1000°F, 4 Hrs.	1350°F, 30 Min.	1725°F, 10 Min.	1900°F, 10 Min.	1000°F, 4 Hrs.		1350°F, Coloria
'A, MANU	Alloy	Class Prote	- 79	4	4	8-1-11	8-1-11	8-1-11	641	- 49		6.4 1	4	<u>4</u>	<u>1</u>			
MIPABILITY DAT	Test	Materia	Single U-Bend	Single U-3end	Single U-Bend	U-Bend	U-Bend	U-Bend	Single U-Bend	Single U-Bend		Flat Sheet	Exposure	_		Allison Bend		
TABLE 4CO	Material		LF.70 (Lubri-Film, Inc.)	LF-75 (Lubri-Film Inc.)	LF-80 (Lubri-Film, Inc.)	LF-500 (Lubri-Film, Inc.)	LF.600 (Lubri-Film, Inc.)	LF-700 (Lubri-Film Inc.)	379-40C (Gold) (Turco Products)	379-40D (Red) (Turco Products)	Turco Pretreat							
	N. Number		27.1.20	27.1.21	27.1.22	27.1.23	27.1.24	27.1.25	27.1.26	27.1.27	27.1.28							

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

						100
Material Number	Material	Test	Alloy	Test Parameters	Results-Observations	Compatibility Rating
		Materia	Class Prot	Material Class Protective Coating, Heat Treat		
27.1.28 (cont'd)				1725°F, 10 Min.	Control 28	
<u>.</u>				1900° F., 10 Min.	<u>5</u>	
					Test 10	
		Emittance	6-41	Flat sheet exposure	Metal	
				specimens as above. Sur-	Temp, Removed Emit-	
-	-			face layer removed by	°F tance	
				chemical milling per	Mils per	
				BAC 5842 as indicated	side	
					1000 0.0 0.15	
					1.5 0.15	
					3.2 0.15	
					2.3 0.15	-
					3.9 0.15	
						ε
						-
					2.1 0.15	
						14 1
						-
					2.6 0.15	
27.1.29	Covirgton Cote (Timet Corp.)					×
		Flat Sheet	6.41	1000°F, 4 Hrs.	No surface effect	
		Exposure		1250°F, 4 Hrs.	Frost pattern	
				1725°F, 30 Min.	Specimen warped	
				1900°F, 30 Mm.	Specimen warped	

TABLE 4 .- COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		∢					×		· ×		×		-						÷-			
Results-Observations			No surface effect	No surface effect	No surface effect	pu	SCC and pits		SCC and pits	ble		Polyvinyl chloride base	No visible surface effect		Bend energy, In-Ibs	Control 67 Test 64	No SCC or surface effect			Specimen broke		
Test Parameters	Material Class Protective Coating, Heat Treat	1000	500° F, 48 Hrs.	700°F, 48 Hrs.	1000°F, 4 Hrs.	Material Class Protective Coating, Spinning Compound	1000°E, 4 Hrs.		1000°F, 4 Hrs.	Materia! Class Protective Coating Temporary, Strippable		Company literature	550°F, 3 Hrs. Two heavy	dip coats, dry between	Specimens as above		850°F, 3 Hrs. Two dip	coats (dry between) plus	one undilled coat	850°F, 3 Hrs. Two dip	coats (dry between) plus	one undried coat
Alloy	Class Prote		1 J	641	4	ss Protectiv	8-1-11		8-1-11	s Protective			6-4 1		- 49		4-3-11			64		
Test	Material		Single U-Bend	Single U-Bend	Single U-Bend	Material Cla	U-Bend		U-Bend	Materia! Clas		Chloride	Flat Sheet	Exposure	Allison Bend		Resin Kettle			Resin Kettle		
Material		Gage Cote 6					Warren No. 1	(Warren Chemical Manufacturing Co.)	Warren No. 2 (Warren Chemical)		Fuller 175-K-14 (W.P. Fuller Co.)											
Material Number		27.1.30					27.2.1		27.2.2		27.3.1											

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating				4			A				-	A			∢														
Results-Observations	able	Specimen broke				No SCC or surface effect	100000000000000000000000000000000000000			No SCC or surface effect	No SCC or surface effect	No SCC or surface effect					Polyvinyl chloride base	No visible surface effect		Bend Energy, In-Ibs	ō	Test 73	No SCC or surface effect		No SCC or surface effect		No SCC or surface effect		
Test Parameters	Material Class Protective Coating Temporary, Strippable	850°F, 3 Hrs, Two dip coats (dry between) plus	one undried coat			1000°F, 4 Hrs.	1000		,	1000°F, 4 Hrs.	1000°F, 4 Hrs.	450°F, 24 Hrs.					Company literature	550°F, 3 Hrs. Two heavy	coats, dry between	Specimens from above		c	850°F, 4 Hrs. One dip	coat, dry	850°F, 4 Hrs. One dip	coat, dry	850°F, 4 Hrs. One dip	coat, dry	
Alloy	S Protective	8.1.1				6-41	5			6-4	8-1-11	9-4						6.4 (6-4 1			4.3.11	_	6.4 1		8-1-11		
Test	Material Clas	Resin Kettle				U-Bend	0.00			U-Bend	U-Bend	U-Bend					Chłoride	Flat Sheet	Exposure	Allison Bend			Resin Kettle-1		Resin Kettle-1		Resin Kettle-1		
Material				1.1043	(Organocerams, Inc.)		1.1044	(Organocerams,	Inc.)			1.2020	(Organocerams,	Inc.)	SC-277-2	(Spraylat Corp.)													
Material Number		27.3.1 (cont'd)		27.3.2			27.3.3					27.3.4			27.3.5								-						

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility	Rating						2	Ŧ									A															
Docules Obsessing	Hesults-Ubservations	able	No SCC or surface effect				No SCC or surface effect					No SCC or surface effect									Bend Energy, In-Ibs	Control 67	Test 68	No SCC or surface effect		No SCC or surface effect		No SCC or surface effect			-	
and a second of the second of	l est Parameters	Material Class Protective Coating Temporary, Strippable	Specimen from Resin Kertle-1 rest left in	kettle, 35 ml of SC-277-2	added, and then heated	at 850°F for 30 Min.	Specimen from Resin	Kettle-1 test left in	kettle, 35 ml of SC-277-2	added, and then heated	at 850°F for 30 Min.	Specimen from Resin	Kettle-1 test left in	kettle, 35 ml of SC-277-2	added, and then heated	at 850°F for 30 Min.			550°F, 3 Hrs. Two heavy	dip coats, dry between	Specimens from above	flat sheet exposure	•	850°F, 4 Hrs. One dip	coat (dry)	850 F, 4 Hrs. One dip	coat (dry)	850°F, 4 Hrs. One dip	cost (dry)			
	Alloy	s Protective	4.3.11				<u>4</u>					8-1-11							6-4 1		Ī			4.3.1 (6-4		8.1.11	-	•		
H	lest	Material Clas	Resin Kettle-2				Resin Kettle-2					Resin Kettle-2							Flat Sheet	Exposure	Allison Bend			Resin Kettle		Resin Kettle		Resin Kettle				
	Materia																SC-1071	(Spraylat Corp.)														
Material	Number		27.3.5 (cont'd)														27.3.6															

TABLE 4 - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Material Number	Material	Test	Alloy	Test Parameters	Results-Observations	Compatibility Rating
			Material C	Material Class Salt, Inorganic		
28.1.1	Silver Chloride (Paste)	U.Bend	6-4 1	1000°F, 4 Hrs.	SCC, Level 5	×
28.1.2	Silver iodide (Paste)	U-Bend	6-41	1000°F, 4 Hrs.	Severe dissolution	×
28.1.3	Silver nitrate (3.5% solution)	U-Bend	6-4	1000°F, 4 Hrs.	SCC, Level 1 to 2	×
28.1.4	Sodium chloride	U.Bend	6.4	1000°F, 4 Hrs. Solution of NaCl in distilled water*	tration SCC fracture, ppm Mils 5 1.8 10 5.2 20 5.6 40 21.3 See photomicrographs in Section 5.4	
29.1.1	8224 (Brandt Chemical Co.)	Mater Double U-Bend	ial Class So	Material Class Scale Conditioner, Aqueous d 6-4 t 265°F, 72 Hrs.	Slight surface polishing No pitting or etching Electron micrographs reveal no attack	∢
29.1.2		Flat Sheet Exposure	6.4	270°F, 1 Hr.	Slight conversion coating	A (operating tem- perature of 270° - 290°F)
The SCC esame CI	"The SCC effect of a given CI concentration in distilled water is greater than same CI concentration in natural water, so that this is a worst-case exposure.	oncentration in distilleral water, so that this i	ed water is g	concentration in distilled water is greater than that of the tural water, so that this is a worst-case exposure.		

TABLE 4 - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating			-	A (operating tem- perature 270 ⁰ . 290 ⁰ F)		-			A (operating tem- perature 190 ^o . 210 ^o F)						
Results-Observations		Bend Energy, In-lbs Control 104	- -		Slight conversion coating	Bend Energy, In-lbs	Control 105 Test 88	Control 105 Test 103		Slight conversion coating	Bend Energy, In-lbs Control 105	Test 77 Control 105	Test 98		
Test Parameters	Material Class Scale Conditioner, Aqueous	270°F, 1 Hr.	270°F, 1 Hr. plus 4 Min. pickle per BAC 5753		270°F, 1 Hr.	270°F, 1 Hr.		270°F, 1 Hr. plus 4 Min. pickle per BAC 5753		195°F, 30 Min.	195°F, 30 Min.	195°F, 30 Min. plus 4	Min. pickle per BAC 5753		
Alloy	ial Class Sca	- 4-9	64 i		- 49	641		- 64		6-4	6-4	6-4			
Test	Mater	Allison Bend	Allison Bend		Flat Sheet Exposure	Allison Bend				Flat Sheet	Allison Bend				
Material				Turco 4316 (Turco Products, Inc.)					Turco 4338 (Turco Products, Inc.)						
Material Number		29.1.2 (cont'd)		29.1.3					29.1.4						

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TABLE 4.—COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility	Rating		A (Immersion time limited to 2 · 5	minutes, with a	30 min. maximum	cumulative)								×		-					-					
	ations	_		-			H2 O2 Surface Pick- Pick- Rough-	990	RHR	13	12	12			O ₂ Surface	Pick- Pick- Rough-	ness	RHR	40	37	31		-	•		
	Results-Observations						Pick 2	=	do moo	-15	115	8			02	Pick-	9	mdd mdd	88	596 240	460					
	esults (H HZ		_	2	18	30			<u> </u>		g	mad	74	969	3102 460					
ľ	ě.						Metal	Ž.		0.0	0.5	0.7	_		Metal	Loss	Mis		8	0.5	3.0				 	
	Test Parameters	Material Class Scale Conditioner, Molton Salt								0.5 hr	2 hr	24 hr							0.5 hr	2 hr	24 hr					
,	Test P	le Conditione								700°F	700°F	700°F						,	500°F	500°F	500°F			•		
	Alloy	al Class Sca								64	6-4	6-4							6.4 1	(Heat	(9.7791)				 	
	Test	Materi					Flat Sheet	3000							Fiat Sheet	Exposure										
	Material		Proprietary (Boeing Manu.	Res.)										Carus 303 (Carus Chem, Co. Inc.)								-				
Material	Number		29.2.1											29.2.2												

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		A (Immersion time	limited to 2 - 5	minutes, with 30	minute cumulative	maximum)			-		-			A (Immersion time limited to 2 · 5 min. with 30 min. cumulative maximum)
Results-Observations				-			Surface	Pick- Pick- Rough-	ness	33		63	210	O2 Surface Pick- Rough- up ness ppm RHR30 80 98 160
bserv							02	Pick	g	ррт 425		327	270	02 up ppm -30 20
Sults-C							H2	Pick.	dn	ppm 21		32	482	H2 Pick up 19 19 57 57 57 57 57 57 57 57 57 57 57 57 57
Re							Metal	Loss	Mils	0.1		8.0	3.7	Metal Loss Mils 0.6 1.7
ameters	Molton Salt									0.5 hr		2 hr	24 hr	0.5 hr 2 hr 24 hr
Test Parameters	Material Class Scale Conditioner, Molton Salt									850°F		850°F	850°F	850°F 850°F 850°F
Alloy	I Class Scal									6-41		6-4 (- 6-4	6-4 6-4 6-4
Test	Materia						Flat Sheet	Exposure						Flat Sheet Exposure
Material		Virgo (Hooker	Chem. Co.)											DGS (Kolene Co.)
Material Number		29.2.3												29.2.4

TABLE 4 - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Material	Test	Alloy	Test Parameters	Results-Observations	Compatibility Rating
		Materi	Material Class Solvent		
Acetic Acid, Glavial				(See Vol. 1, ref 2 for additional data)	A (850°F max.)
	Resin Kettle	4.3.1.1	850°F, 8 Hrs.	No SCC or surface effect	
	Resin Kettle	6-41	850°F, 8 Hrs.	No SCC or surface effect	
Ammonium Hv-	טפאוו עפווופ	1.1.0	000 1, 0 113.	ואס סכר סו מתושב בווברו	×
droxide (29%					
solution)	Resin Kettle	4.3.1.1	450°F, 16 Hrs.	Slight surface effect	
	Resin Kettle Resin Kettle	6.4 8.1.1	450°F, 16 Hrs. 450°F, 16 Hrs.	Slight surface effect Slight surface effect	
Butyl alcohol					A (ambient tem-
	Face H	7.9	75°F 168 Hrs	No CC or curtains affairs	perature only)
		Timet			
		Ht D8839			
		Comp. C			
		Ann		-	
		Type 3			
	U-Bend	8-1-1,	75 F, 168 Hrs. specimens	No SCC or surface effect	
			Method 2		
		Anneal			
	U-Bend	8-1-1,	75°F, 168 Hrs, specimens	No SCC or surface effect	
		Ht D8141	cleaned as above and		
		ΞΞ	heated ½ hr at 1100°F		
		Anneal			
					•

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating		A (850°F max.)		, ***			A				X (Prohibited for	Boeing manu. use due to toxicity)	×						
Results-Observations			No SCC or surface effect	No SCC or surface effect No SCC or surface effect			See 20.1.3. Lubricant/	Coolant						I specimen fractured (SCC) in 29 Hr. 2 speci-	mens showed no SCC	or surface effect			
Test Parameters	Material Class Solvent		850°F, 2 Hrs.	850°F, 2 Hrs. 75°F, 168 Hrs. Speci-	mens cleaned per	BAC 5753, method 2								/5 F, 168 Hrs. Speci- mens cleaned per	BAC 5753, method 2				
Alloy	Materia		6-4	8-1-1-8	Timet,	Ht D8141 Mill	VIII LEGI	•						5-4, Timet	Ht D8839	Comp C	Ann, Tyne 3		
Test			Resin Kettle	U-Bend										U-Bend					
Material		Butyl Carbitol (Diethylene glycol monobutyl ether)	Corp.)				Butvi Cellosolve	(Ethylene glycol	monobutyl ether)	(Union Carbide Corp.)	Carbon Tetra	chloride	Ethanol, absolute						
Material Number		30.1.4					30.1.5	<u>-</u>			30.1.6		30.1.7						

TABLE 4 .- COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility Rating											×																		h y	-	
Results-Observations		1 specimen fractured	(SCC) in 5 Hrs. 1 speci	men showed no SCC or	surface effect	No SCC or surface effect					;	No SCC or surface effect						No SCC or surface effect					No SCC or surface effect	on 3 specimens, 1 speci-	men each fractured (SCC)	in 0.5 and 20 Hrs.			•	-	new .
Test Parameters	Material Class Solvent	75°F, 168 Hrs. Cleaned	as above			75°F, 168 Hrs. Tested in	as received surface	condition			-0	75°F, 168 Hrs. 6 speci-	mens cleaned per	BAC 5753, method 2				As above 3 specimens					75°F, 168 Hrs. 5 speci-	mens, as received plus	0.5 Hr. at 1100°F						
Alloy	Materia	8-1-1,	Timet	Mill Mill	Anneal	8-1-1,	Timet	Ht D8141	Ξį	Anneal	,	6.4	Timet	Ht D8839	СотрС	Ann	Type 3	8.1.1,	Timet	Ht D8141	ΞiΣ	Anneal	8.1.1,	RMI	Sht 504	Ht 30526	Spec MST	8-1-1,	GR470	E W	Anneal
Test						-			•			U-Bend																			
Material										,	Ethylene Glycol																				
Material Number		30.1.7	(cont'd)								30.1.8																_				

TABLE 4 .- COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility	Rating		∢	۷		-							-					A (500°F max.)													
Total Daramente Barrille Observations	nesults-Oose valious		No surface effect. $t = 1.1$		No SCC or surface effect						No SCC or surface effect			No SCC or surface effect						No SCC or surface effect	No SCC or surface effect	No CCC or surface effect	No SCC or surface effect		No SCC or surface effect		3 specimens no SCC or	surface effect 1 speci-	men fracturad (SCC)	No SCC or surface effect	
Tace Decomposite	lest ratameters	Material Class Solvent	500°F, 0.25 Hr.		75°F, 168 Hrs. 1 speci-	men, as received, plus	notch across stressed	area			75°F, 168 Hrs. 4 speci-	mens, cleaned per	BAC 5753, method 2	75°F, 168 Hrs. 2 speci-	mens cleaned per	BAC 5753, method 2	plus 0.5 Hr. at 1100°F		(500 F, 100 Hrs.	500 F, 100 Hrs.	500°F, 156 Hrs.	600°F, 23 Firs, 4 speci-	mens	600°F, 25 Hrs. 4 speci-	mens	600°F, 25 Hrs. 4 speci-	mens	(700°F, 17 Hrs. 4 speci-	mens
)	Alloy	Material I	6-4 1		6-4,	Timet	Ht D8839	Comp C	Ann	Type 3	8-1-1,	Timet	Ht D8141	8-1-1,	Timet	Ht D8141				4-3-11	6-4	8.1.1	4.3.11		- 4-0		8-1-1-1			5.4 1	
• • •	ıest		Heimerl Braski		U-Bend						U -Bend			U-Bend						Res.n Kettle	Resin Kettle	Resin Kettle	Resin Kettle		Resin Kettle		Resin Kettle			Resin Kettle	
	Materia		Freon 1301 (FuPont)	Freon MF (C Cl3F)														Freon PCA	(C Cl2F · C ClF2)										-		
Material	Number		30.1.9	30.1.10														39.1.11													

TABLE 4 - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Material Number	Material	Test	Alloy	Test Parameters	Results-Observations	Compatibility Rating
			Materiz J	Material Class Solvent		
30.1.11			8-1-11	700°F, 17 Hrs. 4 speci-	1 specimen no SCC or	
(cont'd)				mens	surface effect 3 specimens fractured (SCC)	
		Resin Kettle	4.3.11	850°F, 2 Hrs.	Fracture (SCC)	
		Resin Kettle	5-2.5	850°F, 2 Hrs.	Fracture (SCC)	
		Resin Kettle	6.4 1	850°F, 2 Hrs.	Fracture (SCC)	
		Resin Kettle	8-1-1 1	850°F, 2 Hrs.	Fracture (SCC)	
30.1.12	Hexane					٨
		U-Bend	6-4,	75°F, 168 Hrs. Speci-	No SCC or surface effect	
			Timet	mens cleaned per		
			Ht D8839	BAC 5753, method 2		
			Comp C			
			Ann			
			Type 3			
			8-1-1,	75°F, 168 Hrs. Cleaned	No SCC or surface effect	
			Timet	per BAC 5753, method		
			Ht D8141	2 plus 0.5 Hr. at 1100 F		
30.1.13	Methanol		_	(×
		U-Bend	CP 75A	75°F, 51-75 Hrs. Cleaned	Fracture (SCC)	
			Sht 1310	per BAC 5753, method		
			0.050"	2, 3 specimens		
			thick	(
			4-3-1,	75°F, 3.1-17 Hrs. 7 speci-	Fracture (SCC)	
			Crucible,	mens cleaned as above		
			Ht P5946			•
			Ann			
			Type 3			
			0.050"			
		-	gage			
				T		

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Compatibility	Rating				- :					-					-																				
	Results-Observations			Fracture (SCC)						Fracture (SCC)					Fracture (SCC) average	time = 15.8 Hrs.					Fracture (SCC) average	time to fracture = 1.3 Hr.					•	Fracture (SCC) time	to fracture 1.8-8.3 Hrs.	average - 5.5 Hrs.					
	Test Parameters	Material Class Solvent	,	75°F, 1.6-6 Hrs. 6 speci-	mens cleaned per	BAC 5753, method 2				75°F, 4.5·16.5 Hrs. 4	specimens cleaned per	BAC 5753, method 2			75°F, 4 specimens	standard cleaning					75°F, 22 specimens						,	75 F, 11 specimens	standard cleaning						
	Alloy	Materia		5-2.5,	Timet	Sht 55652	D8832	0.050	age	6.4	Timet	ST A8364	0.040.	2686	6-4	Timet	STA	A-8911	0.040	abeb	8:1:1	Timet,	Ht D8141	E.	Anneal	0.040"	Gage	6-4	Timet	Ht 08839	Comp C	Ann	Type 3	0.040"	age
	Test																								_										
	Material				-							-			P-g																				
Material	Number			30.113	(cont'd)																								-						

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

್ ಕರ್ನಾಟಕಾರ್ಯಕ್ಕೆ	Compatibility Rating										:	÷.				A (700°F max.)				A tention	temp. only)							
S (continued)	Results-Observations		Hours to	specimen	fracture	(335)	9	6.5	ω	6		14.5		> 168, no	SCC or		No SCC or surface effect	No SCC or surface effect	(00)	()		No SCC or surface effect			No SCC or surface effect		-	
ATERIALS	Results-C			angle	between	legs,	0	တို့	12°	15	23. 30°	34,0	3/ 48°	e7°			No SCC or	No SCC or	Fracture (SCC)	Fracture (SCC)		No SCC or			No SCC or			
E 4 COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)	Test Parameters	Material Class Solvent	75°F, specimens cleaned,	bolted to produce vary.	ing angles between legs	of Urband, I specimen for each andle (See	figure 2 for plot of data)										700°F, 8 Hrs.	700°F, 8 Hrs.	850°F, 8 Hrs.	850 F, 8 Hrs.		75°F, 168 Hrs. Speci-	mens in as received	condition	75°F, 168 Hrs. Speci-	mens cleaned, plus	1100 F, 0.5 Hr.	
ATA, MAI	Alloy	Materia	6.4	Timet	Ht D8839	Comp C	Type 3	0.040"	aɓeɓ								6-4 1	8-1-1-1	641	- - - - -	_	8-1-11,	Timet	Ht D8141	8-1-11,	Timet	Ht D8141	
MPATIBILITY D	Test																sin Kettle					U-Bend						
TABLE 4CC	Material													- •		Methyl Chloroform	2			Methy! Hiby!	Ketone (NEK)							
	Material Number		30.1.13	(cont.d)										-		30.1.14				30 1 15	?		_					

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Mater.al	Material	Test	Allov	Test Parameters	Results-Observations	Compatibility
Number	0.000	1631				n ating
			Materia	Material Class Solvent		
30.1.16	Perchlorethylene					A (600°F max.)
	(CI2C C CI2)	Bosin Kettle	641	600°F 6 Hrs	No SCC or surface effect	•
			8.1.1	600°F, 6 Hrs.	No SCC or surface effect	
			4-3.11	700°F, 2 Hrs.	1 specimen fractured	
					(SCC) 1 specimen no	-
					SCC or surface effect	,
			6-4	700°F, 2 Hrs. 3 speci-	No SCC or surface effect	
				mens		
			6.4	700°F, 2 Hrs. 5 speci-	5 specimens fractured	
				mens (duplicate run)	(SCC)	
			8-1-1	700°F, 2 Hrs. 1 speci-	Fracture (SCC)	-
30 1.17	n-Propy! Afcohol			men		A (ambient
	(rrobanoi)	U-Bend	8-1-11	75°F, 168 Hrs. Standard	No SCC or surface effect	Certify. Office
			Timet	cleaning		
			Ht D8141			
			8-1-1-	75°F, 168 Hrs. Standard	No SCC or surface effect	
			Timet	cleaning plus 1100°F,		
			Ht D8141	0.5 Hr.		
30.1.18	iso-Propyl Alcohol					A (ambient temp, only)
		U-Bend	6-4 1	75°F, 168 Hrs. Standard	No SCC or surface effect	
			Timet	cleaning		
			Ht D8839			
			Comp C			
			Type 3			
			81-1	75°F, 168 Hrs. Standard	No SCC or surface effect	
			Timet,			
			Ht U8141			
					Ţ	A

TABLE 4. - COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

mersed t value not available but led to A rating (t < 3.71) No SCC or surface effect Fracture (SCC)	Tast Alloy Test Parameters
nmersed t value not available but led to A rating (t <3.71) nmersed No SCC or surface effect osed to No SCC or surface effect Fracture (SCC)	Material Class Solvent
nmersed t value not available but led to A rating (t < 3.71) No SCC or surface effect Fracture (SCC)	
nemersed No SCC or surface effect osed to No SCC or surface effect Fracture (SCC)	Heimerl-Braski 6-4 i 250°F, 65 Hrs. Immersed
No SCC or surface effect Fracture (SCC)	
No SCC or surface effect racture (SCC) Fract	8-1-11 350 F, 12 F
No SCC or surface effect Fracture (SCC) Fracture (S	refluxed glycol
No SCC or surface effect Fracture (SCC) Fractu	Resin Kettle 8-1-11 500°F, 11 Hrs.
No SCC or surface effect Fracture (SCC) Fractu	
No SCC or surface effect Fracture (SCC) F	Resin Kettle 4.3.11 500°F, 100 Hrs.
No SCC or surface effect No SCC or surface effect No SCC or surface effect Fracture (SCC) Fractu	6-4 I 500°F, 100
No SCC or surface effect No SCC or surface effect Fracture (SCC) Fracture (SCC) No SCC or surface effect Fracture (SCC) Fractu	8-1-1 500°F, 10
No SCC or surface effect Fracture (SCC) Fracture (SCC) No SCC or surface effect Fracture (SCC) F	4.3.1. 600°F, 100 Hrs.
Fracture (SCC) Fracture (SCC) Fracture (SCC) No SCC or surface effect Fracture (SCC) Fracture (S	6-4 I 600°F, 10
Hrs. Fracture (SCC) Hrs. Fracture (SCC) No SCC or surface effect Fracture (SCC) Fracture (SCC) Fracture (SCC) Fracture (SCC) Fracture (SCC) No visible SCC or surface effect No SCC or surface effect of Fracture (SCC) No SCC or surface effect of recorded, No SCC or surface effect	_
Hrs. Fracture (SCC) No SCC or surface effect Fracture (SCC) Fracture (SCC) Fracture (SCC) Fracture (SCC) A. Vapor No visible SCC or surface 1AC 5408 effect O^5F, 4 Hrs. No SCC or surface effect of recorded, No SCC or surface effect	6-4 I 700 F, 1000 Hrs.
No SCC or surface effect Fracture (SCC) Fracture (SCC) Fracture (SCC) Fracture (SCC) No visible SCC or surface effect AC 5408 effect No SCC or surface effect or recorded, No SCC or surface effect	
Fracture (SCC) Fracture (SCC) Fracture (SCC) No visible SCC or surface effect No SCC or surface effect No SCC or surface effect or recorded, No SCC or surface effect	ď
Fracture (SCC) Fracture (SCC) No visible SCC or surface effect O°F, 4 Hrs. No SCC or surface effect ot recorded, No SCC or surface effect	_
Fracture (SCC) No visible SCC or surface effect O°F, 4 Hrs. No SCC or surface effect ot recorded, No SCC or surface effect	
s. No SCC or surface effect d, No SCC or surface effect	8-1-11 850°F, 2 Hrs.
s. No SCC or surface effect d, No SCC or surface effect	_
No SCC or surface effect No SCC or surface effect	
No SCC or surface effect	Ht D8141 Now heat 1
ster	8.1.11 75°F. Tin
	Timet distilled water

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

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Materia! Number	Material	Test	Alloy	Test Parameters	Results-Observations	Compatibility Rating
			Materia	Material Class Solvent		
36.1.21			6.4	1000°F, 4 Hrs. Distilled	Slight surface etch	
(cont.d)				water	Clinto confine	
				water	סוולווו אחנוקרב בורוו	
			6-4	1000°F, 4 Hrs. Seattle	Slight surface etch to	
				City water	slight SCC	
			8.1.1	1000°F, 4 Hrs. Seattle	Slight surface etch to	
				City water	slight SCC	
		×	aterial Class	Material Class Stabilizer, Machining		
				,		
31.1.1	Rigidax Blue F (M. Argüeso & Co.,	Double U-Bend	6-4 1	1000°F, 3 Hrs.	Slight etch only	∢
	Inc.)					
31.1.2	Rigidax PF (Argüeso)	Double V-Bend	6-4	1000°F, 3 Hrs.	SCC, Level 5	×
31.1.3	Rigidax W1 Blue	Double U-Bend	6-4	1000°F, 3 Hrs.	Severe etch only	A (?)
31.1.4	Rigidax WI Green	Double U-Bend	6-4 1	1000°F, 3 Hrs.	SCC, Level 5	×
31.1.5	Rigidax W1 NF	Double U-Bend	6-4 1	1000°F, 3 Hrs.	Severe etch only	A (?)
31.1.6	Rigidax YF	Double U-Bend	6-4 (1000°F, 3 Hrs.	SCC, Level 5	×
31.1.7	Barkerwax 3BE	Double U-Bend	1 4 9	850°F, 18 Hrs.	No surface effect	A
	(Barker Enterprises)					
31.1.8	Cerrolow 140	Double U-Bend	64 1	170°F, 16 Hrs.	No surface effect	A
	(Cerrode Pasco	Double U-Bend	6-4	450 F, 16 Hrs.	SCC, Level 5	×
	Corp.) (12.6 Sn,	Double U-Bend	5.4	1000°F, 16 Hrs.	SCC, Level 5	×
	47.5 Bi, 25.4 Pb,					
	9.5 Cd, 5.0 ln)					
31.1.9	Cerrolow 174	Couble U Bend	6.4	48 Hrs.	No surface effect	∢
	(Cerro de Pasco)					
	(5/81, 1/ Sn, 26 In.)					

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Massaria						Compatibility
Number	Material	Test	Alloy	Test Parameters	Results-Observations	Rating
		PM.	nerial Class	Material Class Stabilizer, Machining		
31.1.10	Jarvie 40	Double U-Bend	6.4	650°F, 24 Hrs.	No surface effect	⋖
	(Jarvie Paint Co.)	Double U-Bend	6-4	1000°F, 15 Hrs.	No surface effect	⋖
			Material C	Material Class Stripper, Paint		
32.1.1		Double U-Bend	6-4	75°F, 48 Hrs.	No surface effect	⋖
	(Keelite Corp.)	Double U-Bend	6-4	650°F, 65 Hrs.	SCC, Level 2-3	×
		Double U-Bend	6-4	1000°F, 4 Hrs.	SCC, Level 4	×
32.1.2	Del-Chem 19AC	U-Bend	6-4	450°F, 168 Hrs.	No surface effect	۷.
				550°F, 168 Hrs.	No surface effect	∢
				700°F, 24 Hrs.	Stress corrosion	×
			Material Cla	Material Class Tape, Temporary		
33.1.1	350 (3M Corp.)	Single U-Bend	4 9	350 F, 168 Hrs.	No surface effect	∢
		Single U-Bend	64	500 F, 168 Hrs.	No surface effect	4
		Single U-Bend	6-4	1000°F, 4 Hrs.	No surface effect	A
33.1.2	Y9241 (3M Corp.)	Single U-Bend	6-4 1	350°F, 168 Hrs.	No surface effect	۸.
		Single U-Bend	6-4 1	500°F, 168 Hrs.	No surface effect	• 4
		Single U-Rend	6-4 1	1000°F, 4 Hrs.	SCC, Level 1-2	×
33.1.3	5863 (Mystic Co.)	Single U-Bend	6-4 1	350°F, 168 Hrs.	No surface effect	٨
		Single U-Bend	6-4 1	500°F, 168 Hrs.	No surface effect	⋖
		Single U-Bend	6-41	1000°F, 4 Hrs.	No surface effect	А
33.1.4	6110 (Mystic Co.)	Single U-Bend	6-4	250°F, 168 Hrs.	No surface effect	Α.
		Single U-Bend	6-4 1	1000°F, 4 Hrs.	SCC, Level 1-2	×
33.1.5	733 (Permacel Co.)	Single U-Bend	6-4	350°F, 168 Hrs.	No surface effect	A
		Single U-Bend	6-4	500°F, 168 Hrs.	No surface effect	4
		Single U-Bend	6-4 1	1000°F, 4 Hrs.	Slight surface etch,	٨
- (transfers to a consider seath to book a seafed wheelesses on the contrast the contrast to book a seather than the contrast to	 	l or heat tree	i de di	Level 0	
Remove CC	ereik beiore subse	לרפווו פוופסטופוווא ו	חובשו וובש			

r j

TABLE 4.-COMPATIBILITY DATA, MANUFACTURING AID MATERIALS (continued)

Material Class Tape, Temporary Material Class Tape, Temporary Material Class Tape, Temporary 133.1.7 Single U.Bend 64.1 350°F, 168 Hrs. No surface effect A* 1000°F, 4 Hrs. Single U.Bend 64.1 1000°F, 4 Hrs. No surface effect A 1000°F, 4 Hrs. Single U.Bend 64.1 1000°F, 4 Hrs. Singl	Material Number	Material	Test	Alloy	Test Parameters	Results-Observations	Compatibility Rating
HS 8171 PS Single U-Bend 64 1 350°F, 168 Hrs. No surface effect Single U-Bend 64 1 500°F, 188 Hrs. Single U-Bend 64 1 500°F, 188 Hrs. Single U-Bend 64 1 350°F, 168 Hrs. Single U-Bend 64 1 500°F, 4 Hrs. Single U-Bend 64 1 1000°F, 4 Hrs. Single U				Material Cla	ss Tape, Temporary		
Richmond Corp. Single U-Bend 6-4 500°F, 168 Hrs. No surface effect	33.1.6	HS 8171 PS	Single U-Bend	6-41	350°F, 168 Hrs.	No surface effect	•
Single U-Bend 64 1000°F, 4 Hrs. SCC, Level 1-2 90W (Tuck Single U-Bend 64 350°F, 168 Hrs. No surface effect. Single U-Bend 64 1000°F, 4 Hrs. Slight surface effect, Single U-Bend 64 1000°F, 4 Hrs. Slight surface effect, Level 0 Level 0		(Richmond Corp.)	Single U-Bend	6-4 1	500°F, 168 Hrs.	No surface effect	• •
90W (Tuck Single U-Bend 64 1 350°F, 168 Hrs. No surface effect Single U-Bend 64 1 500°F, 168 Hrs. No surface effect. Single U-Bend 64 1 1000°F, 4 Hrs. Slight surface effect, Level 0 Level 0			Single U-Bend	641	1000 F, 4 Hrs.	SCC, Level 1.2	×
Single U.Bend 6-4 i 500°F, 168 Hrs. No surface effect. Single U.Bend 6-4 i 1000°F, 4 Hrs. Slight surface effect, Level 0 Level 0	33.1.7	90W (Tuck	Single U-Bend	641	350 E, 168 Hrs.	No surface effect	4
Single U-Bend 6-4 i 1000° F, 4 Hrs. Slight surface effect, Level 0 Charles and treatment stress and treatment		Tape Co.)	Single U-Bend	641	500°F, 168 Hrs.	No surface effect	⋖
an stream of heat treatment			Single U-Bend	6-4:	1000°F, 4 Hrs.	Slight surface effect,	∢
D amount of the former of the form of the							
Paramonomidately before risk contents treatment							
Permitted to heart enterentiant strategicality of host trastrant							
Pommun Comulated to Active cultivation of these residuation or has treasment							
* Benny Comilately hefore culted invited treastrain						•	
* Benow comilately before subscaling or host reservalition or host reservalition or host reservalitions							
* Bemove commissed to be fine cultivation of heat treatment							
*Ramous comulately before cubenium creasing							
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* Remove completely before cubronions treatment							
Remove completely before cubeniant streetralianing or hast treatment							
*Ramous comulately before cubeniant stracerelianing that treatment							
*Removes completely before cubeniant straceralisation or host treatment							
*Ramous complete substitution or hast treatment							
*Ramous complete substantiant streetralitation or hast treatment							
*Remove complete teleconiant straceralisation or hast treatment							
* Remove completely before subsequent stressread							
*Remove completely before cubeniant straceraliaving or hast treatment							
*Removes comulately before subsequent strace religioning or heat treatment							
	Bemove	f omulately before subse	 	- Cr heat tres			

5.4 SODIUM CHLORIDE INDUCED STRESS CORROSION CRACKING OF TITANIUM

The stress corrosion produced in Ti-6A1-4V U-bend specimens by varying low concentrations of chloride ion in distilled water is illustrated by the photomicrographs in figures 5 to 12. Exposures were conducted as described in section 3.3.2. One interesting observation was that stress corrosion cracking was more severe with sodium chloride dissolved in distilled water than with a comparable solution of sodium chloride in tap water.

5.5 METHANOL INDUCED STRESS CORROSION CRACKING OF TITANIUM

Special interest in the compatibility of methyl alcohol with titanium alloys was prompted by the failure of an Apollo fuel tank in October 1966, during pressure testing with methanol. Methanol test data obtained in this program is given in table 4, material code number 30.1.13. This section contains metallographic results on several fractured specimens.

Fractographic examinations were carried out on three specimens: No. 8-270, Ti-8Al-1Mo-1V; No. 8-39, 8Al-1Mo-1V; and No. 6-73, Ti-6Al-4V. Each specimen was a standard U-bend, and immersed in methanol containing 400 ppm of water until failure. One similar specimen, immersed in methanol containing 1300 ppm of water, did not fail in two weeks, but failed within two hours when transferred into methanol of low water content.

One half of each specimen was cut and mounted for metallographic examination, and the opposite half retained intact for electron microscopic examination. The metallographic profiles of the secondary cracks showed a transgranular crack path (figures 14-20). The Ti-6A1-4V specimen no. 6-73 showed that the beta phase is more ductile than the alpha phase.

Replicas of the origin which represents the slow crack growth region always exhibited mixed cleavage and ductile fracture features. A lesser degree of cleavage was observed in the areas of rapid fracture (see figures 21-23). The difference in fractographic features between the origin (slow growth region) and the areas of rapid fracture was not as profound as expected. The microscopic difference was more outstanding.

Specimen No. 6-73, Ti-6A1-4V did not exhibit any well defined slow growth regions. Representative areas showed mixtures of cleavage and ductile fracture features (figure 23).

All the specimens that have failed when immersed in methanol, both Ti-6A1-4V and Ti-8A-1Mo-1V, did so in a brittle fashion with cleavage features typical of stress corrosion cracking. The features are similar to those found in specimens that have failed in 3.5 percent salt water environments. The beta or transformed beta regions behave in a ductile manner. Williams (ref. 11) has found both alloys, Ti-6A1-4V and Ti-8A-1Mo-1V, generally exhibit cleavage failure through the primary alpha phase, and ductile features through the beta and transformed beta regions during stress corrosion.

Beck and Blackburn (ref. 12) found that in salt water environments the fracture mode of the alpha phase usually changes from cleavage in the SCC zone to ductile rupture in the rapid fracture zone. This change was not observed in these specimens. The difference is believed to be due, in part, to the difference in specimen size and geometry. Most experimental work has been conducted using heavier gage, notched bend specimens with the crack traveling in the long, transverse direction whereas the subject specimens cracked in the short transverse direction.



FIGURE 5.—HOT-SALT SCC OF TI-6AL-4V (5 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS)

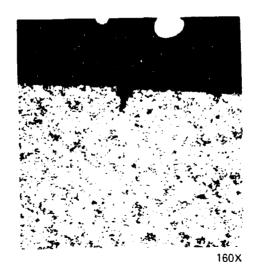


FIGURE 6.-HOT-SALT SCC OF TI-6AL-4V (5 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS) DEPTH OF FRACTURE, 1.8 MILS



FIGURE 7.—HOT-SALT SCC OF TI-6AL-4V (10 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS)



190X

FIGURE 8.—HOT-SALT SCC OF TI-6AL-4V (10 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS) DEPTH OF FRACTURE, 5.2 MILS



FIGURE 9.—HOT-SALT SCC OF TI-6AL-4V (20 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS)

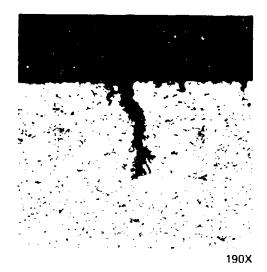


FIGURE 10.—HOT-SALT SCC OF TI-6AL-4V (20 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS) DEPTH OF FRACTURE, 5.6 MILS



FIGURE 11.—HOT-SALT SCC OF TI-6AL-4V (40 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS)

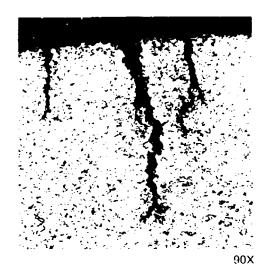
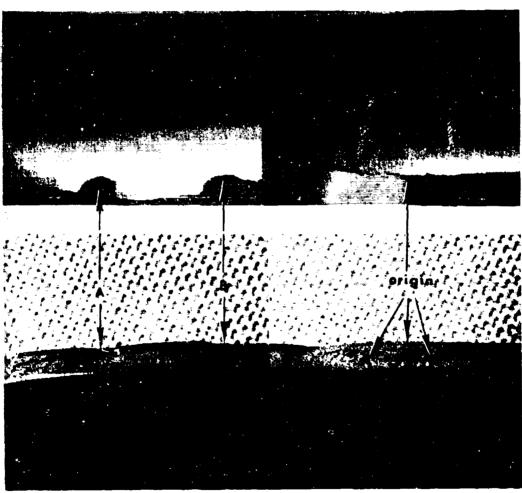


FIGURE 12.—HOT-SALT SCC OF TI-6AL-4V (40 PPM CL IN DISTILLED WATER; 1000F, 4 HOURS) DEPTH OF FRACTURE, 21.3 MILS

Origin and General Surface Topography
Areas A and B are believed to be Secondary Origins



Ti-8A1-1Mo-1V

FIGURE 13.--PROFILE AND FRACTURE SURFACE OF SPECIMEN NO. 8-270

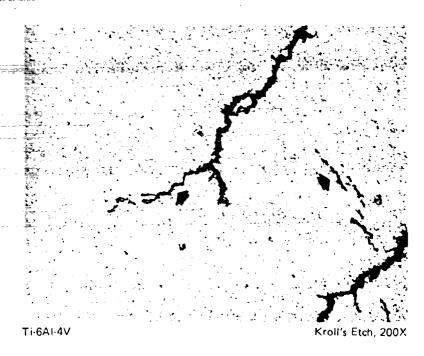


FIGURE 14. -SPECIMEN NO. 6-73 SECONDARY CRACK PROFILE (SHOWING A STEP LIKE CRACK PATH WHICH IS TRANSGRANULAR AT HIGHER MAGNIFICATIONS)



FIGURE 15.—SPECIMEN NO. 6-73 AREA SHOWING THE TRANSGRANULAR NATURE OF THE CRACK PATH

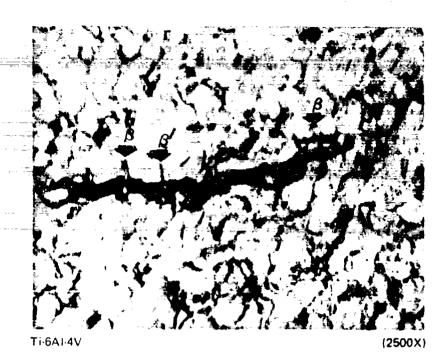


FIGURE 16.—SPECIMEN NO. 6-73 REPRESENTATIVE AREA SHOWING DUCTILITY OF THE BETA PHASE

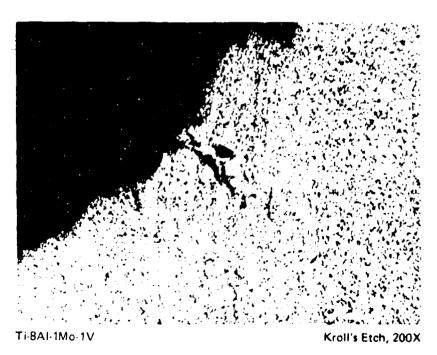


FIGURE 17.--SPECIMEN NO. 8-39 PRIMARY AND SECONDARY CRACK PROFILE SHOWING A CRACK PATH SIMILAR TO TI-6A1-4V



Ti-8AI-1Mo-1V

Kroll's Etch, 2500X

FIGURE 18.—SPECIMEN NO. 8-39 AREA AT A HIGHER MAGNIFICATION SHOWING THE TRANSVERSE CRACK PATH AND SOME ELONGATED PARTICLES



Ti-8AI-1Mo-1V

10,000X

FIGURE 19.—SPECIMEN 8-270 SLOW CRACK GROWTH REGION (ORIGIN) SHOWING A MIXTURE OF PRIMARILY CLEAVAGE AND SOME DUTILE AREAS



FIGURE 20.—SPECIMEN NO. 8-270 SLOW CRACK GROWTH REGION, SIMILAR TO FIG. 19



Ti-8Al-1Mo-1V 10,000X

FIGURE 21.-SPECIMEN NO. 8-270 RAPID CRACK REGION SHOWING MORE DUCTILITY WITH SOME AREAS HAVING CLEAVAGE



Ti-8AI-1Mo-1V 10,000X

FIGURE 22.-SPECIMEN 8-270 RAPID CRACK REGION, SIMILAR TO FIGURE 21

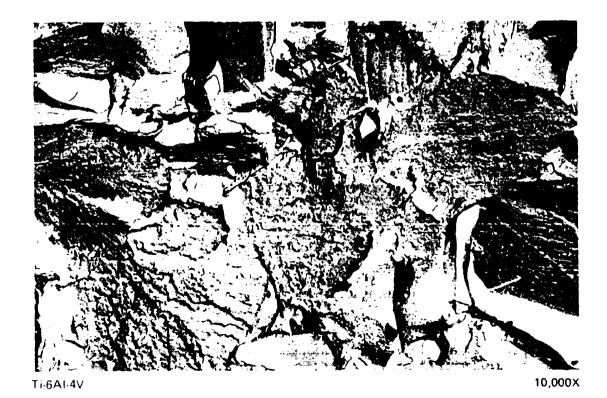


FIGURE 23.—SPECIMEN NO. 6-73 REPRESENTATIVE AREA ON THE TENSION SIDE SHOWING WELL DEFINED CLEAVAGE FEATURES

6.0 DISCUSSION OF RESULTS

The test results in section 5 are very specific, e.g. the data deals with individual materials tested under particular conditions of temperature and time. This is especially true since most of the materials tested were proprietary products for which chemical composition information was lacking. It is difficult to deduce general statements about compatibility of a class of materials, such as heat treat protective coatings, which are used for a single purpose but may include a variety of chemical compositions. In a sense each material studied constitutes an individual test program, with the test data given in table 4 and the test result summarized in the compatibility rating included therein.

6.1 TEST METHODS

The test methods used to evaluate titanium compatibility changed and evolved during the 1967-1971 time period covered in this document. In the earlier work several test methods were used in studying each material. Later, for reasons discussed below, primary reliance was placed on the U-bend specimen, and work was begun on modifying the method by introducing the double U-bend test for faying surface entrapment simulation, and the indented U-bend test for increased sensitivity.

6.1.1 The Modified Allison Bend Test

This test, as modified, was designed for the detection of the formation of brittle surface layers on exposed specimens. That it is capable of doing so is indicated by data for several scale conditioners (29.1.2-29.1.4, table 4). The 15-25 percent decrease in bend energy of exposed specimens compared to controls is correlated with the formation of a conversion coating. Return of the bend energy to its original value following a short pickling process to remove the surface layer confirms that the bend energy decrease is detecting a surface change.

In a similar fashion this test was capable of detecting surface oxidation produced during very high (1725° and 1900°F) temperature exposures while testing heat treat coatings. (See 27.1.6-27.1.9 and 27.1.28.) However, no significant differences were observed between control and test specimens. This suggests that heat treat coatings produce neither deleterious or protective effects on the titanium.

Reviewing the Allison bend test data led to the conclusion that while the method did detect surface changes, it did not appear to be suitable for studying titanium-material incompatibility. This is especially the case since the specimens were exposed to the test material unstressed, so that the occurrence of stress corrosion cannot be detected.

6.1.2 Emittance Testing

This test was useful for studying the formation of oxide coatings formed on titanium by exposure to high temperatures (1725 and 1900°F) during evaluation of heat treat protective coatings (see 27.1.8, 27.1.9, and 27.1.28). The presence of an oxide coating was indicated by

an increase in the emittance of the surface. A measure of the thickness of the coating was obtained by chemically removing the oxide layer, and measuring the removal required to return to the emittance of the bare metal. As expected this test shows that the oxide film formed in ten minutes at 1900°F is thicker than the film formed in ten minutes at 1725°F. This test can be used to measure the protective ability of heat treat coatings. It was not, however, particularly useful in studying material compatibility.

6.1.3 Chlorine Analysis

Initially it had been planned to rely upon the chlorine content of manufacturing aid materials as a criteria for compatibility, with a maximum of 200 ppm allowed. This was based upon the well-documented occurrence of hot salt stress-corrosion cracking of titanium, and the knowledge that even covalent chlorine compounds will form ionic or ionizable decomposition products at high temperatures.

A number of dye penetrant materials were analyzed for chlorine, but no correlation existed between chlorine content and occurrence of corrosion. In the seven out of thirty-five analyzed materials for which corrosion was observed the chlorine content varied from 5-384 ppm. In the remaining 28 materials which did not produce SCC the chlorine content varied from 51 ppm to 1 percent. In particular, sample 26.3.27 at 700 ppm and samples 26.3.29 and 26.3.30 at 1 percent chlorine passed the corrosion test. This may be because the chlorine was contained in some volatile molecule.

For this reason it is recommended that no arbitrary maximum chlorine content be established, but rather that the chlorine analysis be used as a tool in helping to establish the cause of corrosion, and as a warning to carry out especially careful corrosion testing for materials with high chlorine contents.

6.2 RESIDUE REMOVAL

For many manufacturing aid materials the test temperatures used were extremely severe, well above those encountered during their use in the shop, or subsequently in the airplane. These were employed because of the possibility that titanium parts contaminated by these materials might subsequently be heat treated or stress-relieved at such temperatures. Materials rated X in table 4 may well be usable, if required, in manufacturing if adequate provision is made for removal of residual traces of the material following use.

6.3 DISCUSSIONS OF MATERIAL CLASSES

No general discussion of each class of manufacturing aid materials is possible for reasons previously cited, namely that the classes are grouped by function and not by chemical character, and that the test results are specific for each particular material. General comments about a few classes are given below.

In several cases a much larger number of individual materials within a class were examined than were required to select manufacturing aid materials suitable for the stress-corrosion standpoint. This was done to study the test methods in more detail. One major advantage of

the U-bend test is that the extra expense to include additional materials in a test series is quite low. For example a large number of dye penetrant materials were examined, in part to determine whether production of stress corrosion was correlated with chlorine content.

6.3.1 Machining Lubricants and Coolants

Three members of this class were rated X (incompatible) (21.1.8, 21.1.37 and 21.1.41) based upon the presence of chlorinated solvents in their formulation, even though no evidence of corrosion was observed during testing. However all three of these were tested at an early date before the use of U-bend specimens in the resin kettle test was initiated. In view of the stress corrosion observed for lubricant material 21.1.4, and for several chlorinated solvents in resin kettle tests this seems to be a reasonable, conservative conclusion.

6.3.2 Electrochemical Marking Materials

The X ratings assigned to these materials are almost redundant, since their marking functioning depends upon their ability to attack titanium. Possibly they may safely be used for marking purposes if employed as directed, and carefully removed subsequently.

6.3.3 Penetrant Inspection Fluids

A wide variety of all three subclasses were tested, penetrants, developers, and emulsifier-removers. None of the 41 penetrants tested exhibited incompatibility. This is very reassuring since these are formulated to penetrate into small cracks from whence their complete removal may be difficult. While some developers and emulsifiers produced stress-corrosion it was generally moderate. In any case a large variety of compatible systems were discovered.

6.3.4 Solvents

Test data on the different solvents exhibited a good deal of variability. In part this was attributed to differences in surface preparation; for this reason the nature of the specimens is described in considerable detail in table 4. The data on ethanol, 30.1.7, shows the importance of surface preparation in SCC. The data allows several interesting conclusions.

- The greater stress-corrosion susceptibility of Ti-8A1-1Mo-1V compared to Ti-6A1-4V is shown both for alcohols and for chlorinated compounds.
- Chlorinated solvents (and fluorochlorocarbons) produce stress-corrosion cracking
 in the resin kettle, U-bend test at high temperatures, where thermal decomposition
 of the partially confined solvent can occur. It is probable that this effect would not
 have been noted under test conditions such that the solvent vapors could have diffused away from contact with the specimens.



7.0 CONCLUSIONS AND RECOMMENDATIONS

7.1 CONCLUSIONS

- Compatibility ratings, valid for the test conditions listed in the data tabulation, have been determined for 267 manufacturing-aid materials.
- The simple U-bend test is the most useful method for screening the compatibility of materials with titanium.
- For volatile liquids, the resin kettle modification of the U-bend test is recommended.
- Emittance measurements, combined with controlled chemical etching, can be used to measure the extent of surface oxidation.
- The modified Allison bend test can detect surface changes in titanium, especially the formation of oxide coatings. It is not a useful test for monitoring titanium compatibility of materials.
- Halogenated solvent vapors, restrained in contact with titanium alloys, can produce stress corrosion cracking at 600°F and above.
- The hot extraction method is preferable to the vacuum fusion method for measuring the hydrogen content of titanium.
- The fracture depth produced by 1000°F, 4 hour exposure of titanium U-bend specimens to evaporated trace salt solutions is proportional to the salt concentration of the solutions.
- The chlorine content of manufacturing aid materials is not, per se, a reliable criterion of ability to induce titanium corrosion.
- The fracture mode of the specimens that failed in methanol is typical of transgranular stress corrosion cracking. The alpha phase failed by cleavage and the beta phase by ductile rupture.
- The primary fracture origin and secondary origins of Ti-8A1-1Mo-1V show macroscopically distinguishable zones. Microscopically, the fracture mode of the alpha phase does not change as markedly from the stress corrosion cracking zone to the rapid fracture zone.
- The fracture modes for both the Ti-8A1-1Mo-1V and Ti-6A1-4V are similar to those observed in stress corrosion cracking in saltwater environments.
- As the water content increases in the reagent grade methanol, the susceptibility to stress corrosion cracking decreases.

7.2 RECOMMENDATIONS

- Batch testing of materials using the U-bend specimen is recommended as the primary test for titanium compatibility.
- Chlorine content of manufacturing aid materials should be measured and used as supplemental information in interpreting compatibility test results.
- The compatibility ratings listed in this document should be used with care, taking into account the specific tests used in deriving each rating.
- Special manufacturing methods and controls should be developed to ensure com -plete removal of any manufacturing aid materials producing SCC at very high temperatures, prior to subsequent heat treat or stress relief operations.

APPENDIX A

METHODS OF CHLORINE ANALYSIS

The following methods of chloride analysis were utilized. The particular method used depended upon the type of material involved and the accuracy required.

1. BOSLER METHOD

This test was used to determine the amount of chlorides in machining lubricants.

Method I. Combustible (Flammable) Lubricants

Five to ten grams of material was burned and the combustion products were aspirated through a 10% NaOH solution. The halides were precipitated with silver and the amount determined gravimetrically.

Method II. Noncombustible Lubricants

A known weight of sample was taken and boiled with 50% nitric acid and silver nitrate. The resultant solution was filtered and washed, then weighed for determination of the amount of chloride.

Both methods were capable of detecting chloride concentrations of greater than 100 ppm.

2. RADIOTRACER SILVER-110

The material to be analyzed is dissolved in a nitric-hydrofluoric acid solution to which a radiotracer of silver-110 is added to precipitate the chloride. The silver chloride is removed by centrifuging and the residual activity counted with scintillation equipment. The procedure will detect 2 mg of chloride.

3. BEILSTEIN TEST

A copper wire is dipped into the candidate solution and the wire placed in a nonluminous gas of alcohol flame. A characteristic green or blue-green color indicates the presence of halide. The test is qualitative.

4. BOMB TEST, PARR

The methods for chloride analysis as described have been successfully used on materials other than penetrants. The method used for actual precipitation of the chloride has varied for some analyses from that described.

Method I. Determination of Chloride in Penetran's

Apparatus

Oxygen Bomb - Parr or equivalent per ASTM D129-64

Photometer - Klett-Sommerson or equivalent, or a spectrometer, or a

nephelometer

Buret – 5 ml capacity with 0.01 ml divisions

Reagents

Potassium Hydroxide Solution 3N

Dissolve 168 grams of potassium hydroxide (ACS Reagent Grade) in 1 liter of distilled or deionized water.

Nitric Acid Solution 3N

Dilute 195 ml of concentrated nitric acid (ACS Reagent Grade) to 1 liter with distilled or deionized water.

Silver Nitrate Solution 1%

Dissolve 10 grams of silver nitrate (ACS Reagent Grade) in 1 liter of distilled or deionized water.

Potassium Chloride Solution 100 micrograms per ml

Dissolve 0.2103 grams of potassium chloride (ACS Reagent Grade) in distilled or deionized water in a 1-liter volumetric flask and dilute to mark.

Standardization

Using the 5-ml buret, measure carefully into 10-ml volumetric flasks the following quantities of potassium chloride solution: 0.10, 0.20, 0.30, 0.40, 0.50, 0.60, 0.70, 0.80, 0.99, and 1.00 ml. To each of these flasks add 2.0 ml 3N potassium hydroxide solution, 3.0 ml 3N nitric acid solution, 1 ml 1% silver nitrate solution and dilute to mark with distilled or deionized water. Invert once and place into a 40° C constant temperature bath for a minimum of 30 min. Remove from the bath and cool rapidly under running cold water. Within an hour of placing the sample into the constant temperature bath, read the absorbance of the sample versus a blank, prepared by adding similar quantities of the reagents to distilled or deionized water in a 10-ml volumetric flask. With the Klett-Sommerson photometer a No. 42 filter was used. For other instruments, the optimum wavelength would have to be determined by a qualitative scan through the visible region for the point of maximum absorbance. This point should give the maximum sensitivity. Plot a curve of absorbance versus concentration. The above solutions contain 10 through 100 micrograms of chloride, respectively.

Procedure

Weigh to the nearest 0.1 mg approximately 0.8 gram of sample into the crucible for the oxygen bomb. Add 2 ml of 3N potassium hydroxide to the bomb, assemble and add the required quantity of oxygen, Ignite and allow to stand for a minimum of 15 min. Disassemble and wash the contents into a 250-ml beaker with distilled or deionized water. Evaporate the contents of the beaker to approximately 20 ml, then filter through a moderately retentive filter, such as S&S 489

white ribbon, into a 100-ml beaker. Wash thoroughly with hot distilled or deionized water. Fvaporate the contents of the beaker almost to dryness, or to dryness, but do not allow the residuc to bake. Cover the beaker with a watch glass and add 3 ml of 3N nitric acid dropwise, swirling the beaker gently between drops until effervescence ceases. Wash the cover glass and contents of the beaker into a 10-ml volumetric flask with distilled or deionized water and dilute to mark. Pipette 5 ml of the solution into another 10-ml volumetric flask, add 1 ml of silver nitrate, dilute to mark, and place it into a 40°C constant temperature bath for a minimum of 30 min. Remove from the bath and cool rapidly under cold running water. Within an hour of placing the sample into the constant temperature bath, read the absorbence of the solution as compared to a blank prepared by diluting the remaining 5 ml of sample in the other 10-ml volumetric flask to the mark with distilled or deionized water. From the curve prepared under standardization determine the chloride ion concentration in micrograms. The parts per million concentration is then calculated as follows:

 $ppm Cl^{2} = \frac{(Cl^{2} \text{ in micrograms})}{Sample \text{ weight in grams}}$

Method II. Determination of Chloride in Nonoxidizable Materials

Procedure

Weigh a sample of approximately 0.5 gram to the nearest milligram upon a piece of glazed paper. Mix with approximately 2 grams of A.R. sodium carbonate and add to a platinum crucible containing approximately 1 gram of A.R. sodium carbonate. Cover the mixture with another gram of A.R. sodium carbonate and place in a muffle furnace. Let the temperature come up slowly until 850° to 900° C is reached and then allow to fuse for 30 min. Turn off the muffle furnace and allow crucible to cool. Transfer the cold crucible and contents to a 250-ml beaker containing 100 ml of distilled or deionized water to which 5 ml of concentrated nitric acid has been added. Digest upon a hot plate until the fusion mass has been decomposed, then remove the crucible washing thoroughly with hot distilled or deionized water. Filter the solution through a medium retentive filter and wash with hot distilled water into a 250-ml beaker. Evaporate on a hot plate to a volume of approximately 20 ml. If a precipitate forms, refilter and again evaporate. After a clear solution of 20 ml or less is obtained, transfer to a 25-ml volumetric flask and finish the determination by silver chloride turbidity as described in Method I. Adjust the calculation of the silver concentration for the difference in final volumes.

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REFERENCES

- 1. Boeing Document D6A11108-1, Compatibility of Manufacturing Aid Materials with Titanium Alloys, P.S. Jacobsen, July 1968
- 2. Boeing Document D6-60208-1, (FAA-SS-72-08-1), Compatibility of SST Materials with Titanium Alloys: Volume 1, Flyaway Materials, P.S. Jacobsen and A.E. Senear. Released July, 1972.
- 3. Fontana, Mars C., Stress Corrosion of Titanium and its Alloys, Ind. and Eng. Chem. 48, 59A-60A (1956)
- 4. Stough, D.W., Fink, F.W., and Peoples, R.S., *The Stress Corrosion and Pyrophoric Behavior of Titanium and Titanium Alloys*, TML Report No. 84. DMIC, Battelle Memorial Institute, Columbus, Ohio, Sept. 15, 1967
- 5. Braski, D.N., and Heimerl, G.J., The Relative Susceptibility of Four Commercial Titanium Alloys to Salt Stress-Corrosion at 550°F, NASA Technical Note D-2011, Dec. 1963
- 6. Brown, B.F., A New Stress Corrosion Test for High Strength Alloys, Materials Res. and Stand. pg 129-33, March 1966
- 7. Evaluation of Titanium and Titanium Alloys in Alcohols and Halogenated Hydrocarbons, DMIC Tech. Note, Dec. 1, 1966
- 8. F. L. Parkinson, *Titanium Alloy 6Al-4V Hydrogen Effects* FAA-SS-72-07, June 1972.
- 9. Heimerl, George J., and Braski, D.N., A Stress Corrosion Test for Structural Sheet Material, Materials Res. and Stand. pg. 18-22, Jan. 1965
- 10. Hanink, D.K., Metallurgical Tests as a Contribution to Pressure Vessel Reliability, Metal Progress 77, 152-4 (1960)
- 11. J.C. Williams, Some Observations on the Stress Corrosion Cracking of Three Commercial Titanium Alloys, Trans. Am. Soc. Metals 60, 646 (1967)
- 12. T.R. Beck and M.J. Blackburn, Stress Corrosion Cracking of Titanium Alloys: Electrochemical Mass-Transport-Kinetic Model, Metallurgical and Mechanical Effects, and Proposed Relation of Electrochemical, Metallurgical, and Mechanical Effects, Quarterly Progress Rep. No. 4, Ap. 1, 1967—June 30, 1967, Contract NAS 7-489.